

DESIGN OF A MARS POLAR RESEARCH BASE WITH CREW

Anne-Marlene Rüede*, **Anton Ivanov†**,
Claudio Leonardi* & **Tatiana Volkova***

** Ecole Polytechnique Federale de Lausanne, Lausanne, Switzerland*

† Skolkovo Institute of Technology, Moscow, Russia



eSPACE

Space Engineering
Center



“North Polar Layered Deposits in Head of Scarp of Chasma Boreale”
photographed by HiRISE, Mars, 2006
courtesy of NASA/JPL-Caltech/University of Arizona

Mars North Pole

- 1 Determine if Mars ever supported life
- 2 Understanding the processes and history of climate on Mars

- 3 Understand the origin and evolution of Mars as a geological system
- 4 Prepare for human exploration

Objectives

1

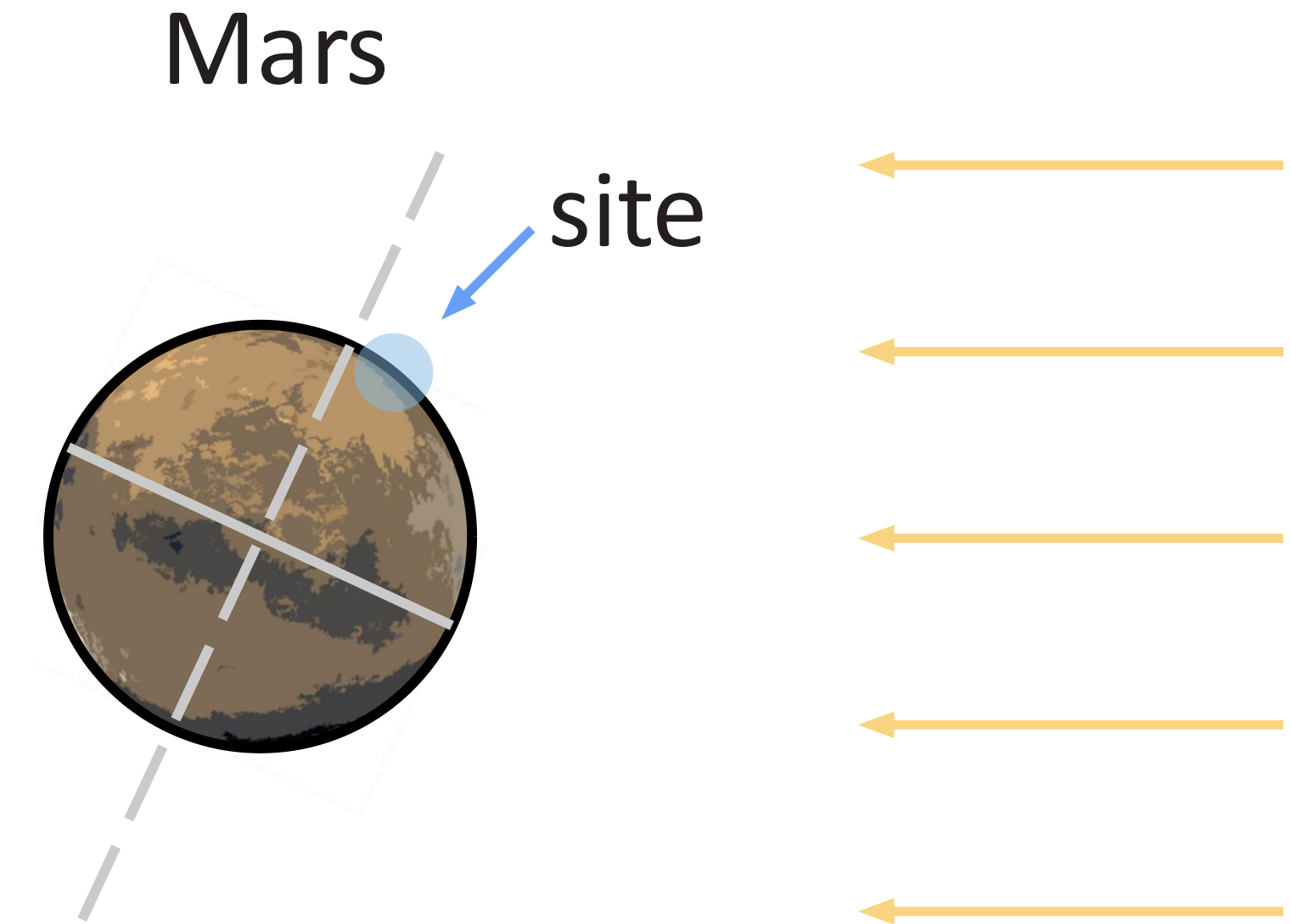
Propose a design for a
9-month manned Mars surface
mission

2

Identify mission design or
technology development areas

Mission Definition

- Human crew
- 288 days surface mission at Mars North Pole during constant daylight
- Drilling & analysis of Polar Layered Deposits (PLD)
- Safe return of the crew on Earth



Crew Composition



piloting &
navigation



electrical &
communication engineer



mechanical
engineer



medecine, dentistry
& psychology



geology & data
processing



biology &
botany

introduction

strategy selection

base design

crane design

conclusions

Resources on Mars



bricks, mortar,
ice, ...



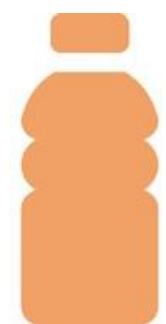
steel, aluminum, silicone



glass, pottery



water, oxygen, nitrogen,
food complements

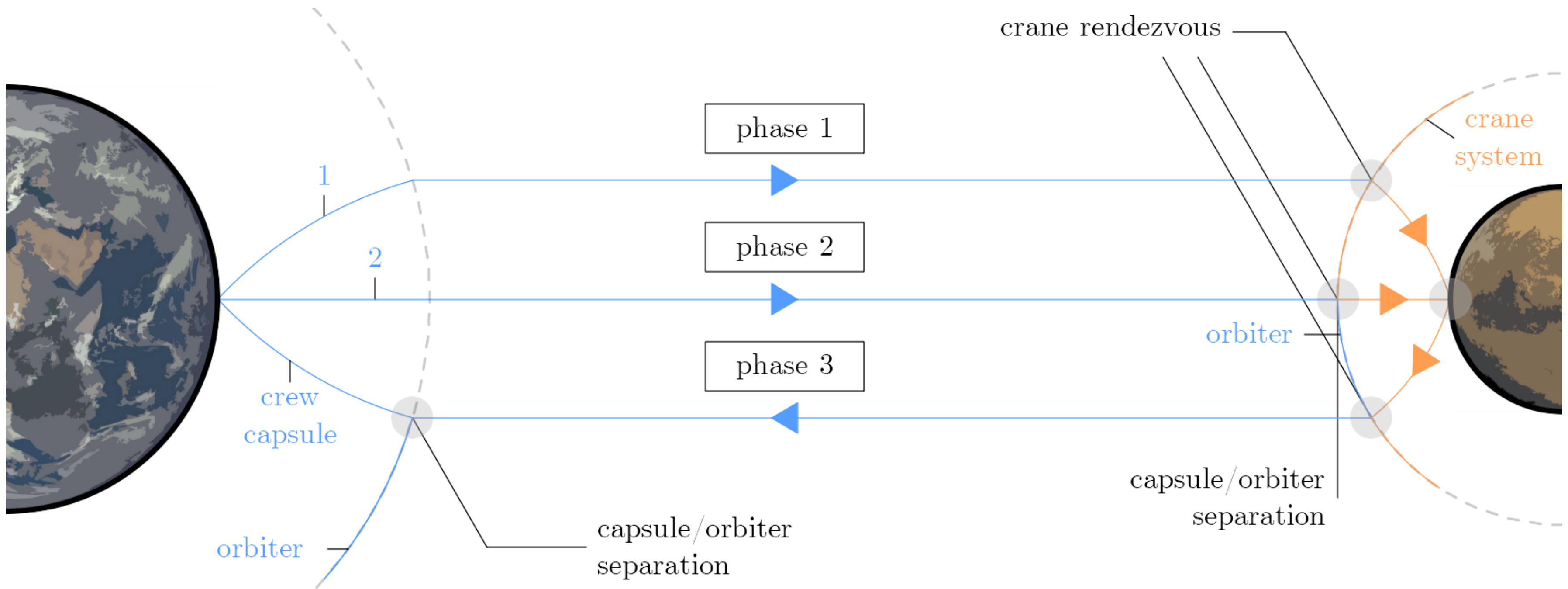


ethylene based
plastics



hydrogen, carbon monoxide,
methane, methanol, ethylene,
ALICE (including oxide)

Mission Phases



introduction

strategy selection

base design

crane design

conclusions

Life Support System



food



energy



water



oxygen



nitrogen

■ materials brought from Earth ■ Mars in-situ collected materials

introduction

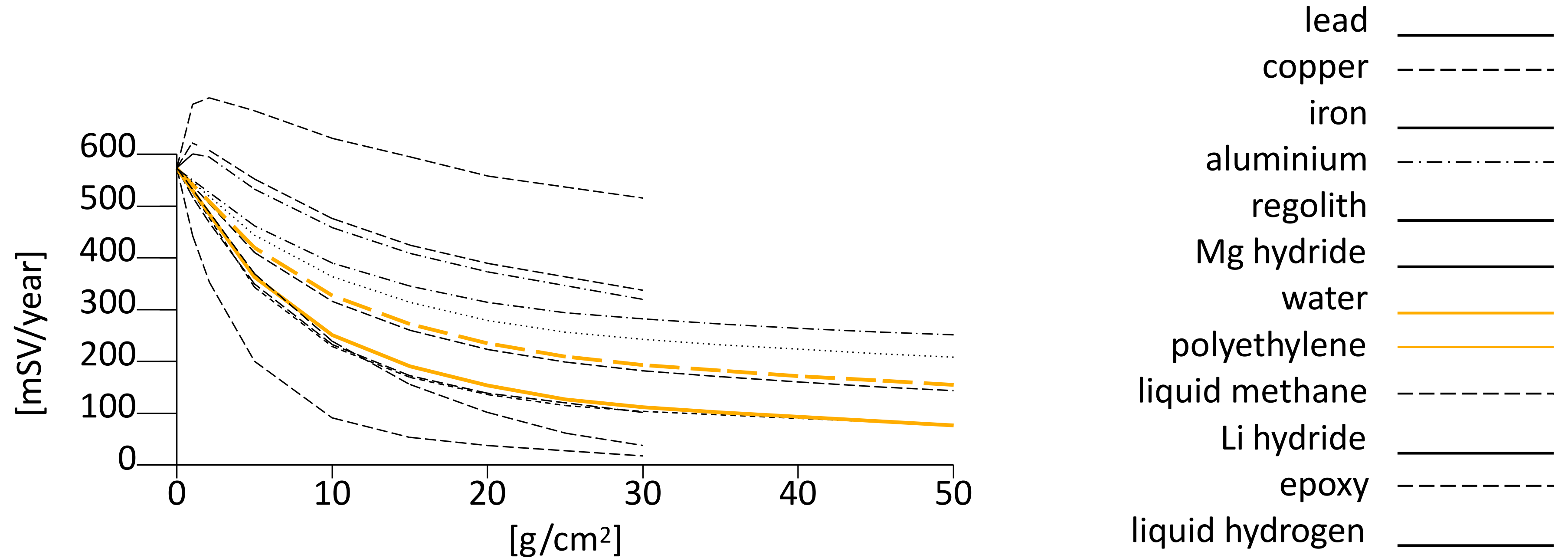
strategy selection

base design

crane design

conclusions

Radiation Protection



introduction

strategy selection

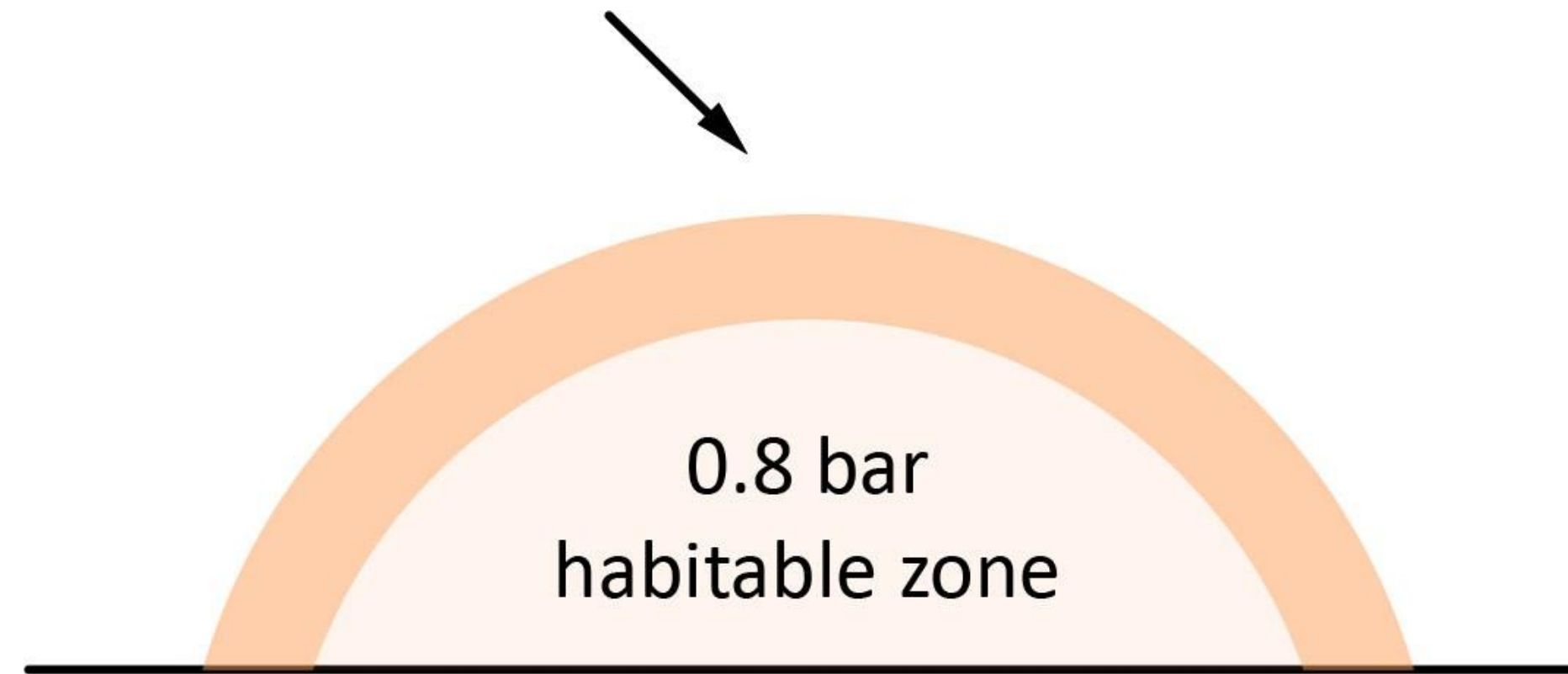
base design

crane design

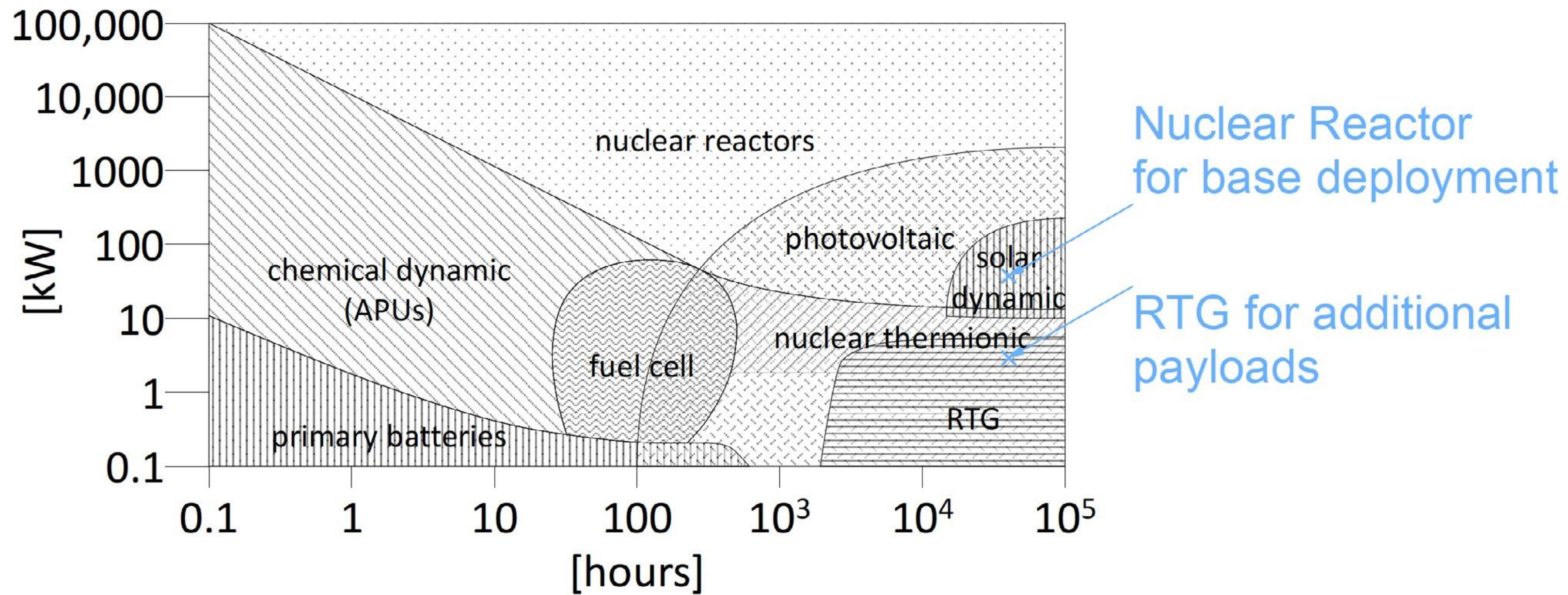
conclusions

Air Pressure

in-situ minned ice as mass to compensate
the pressure difference (also serves as
radiation protection)



Power

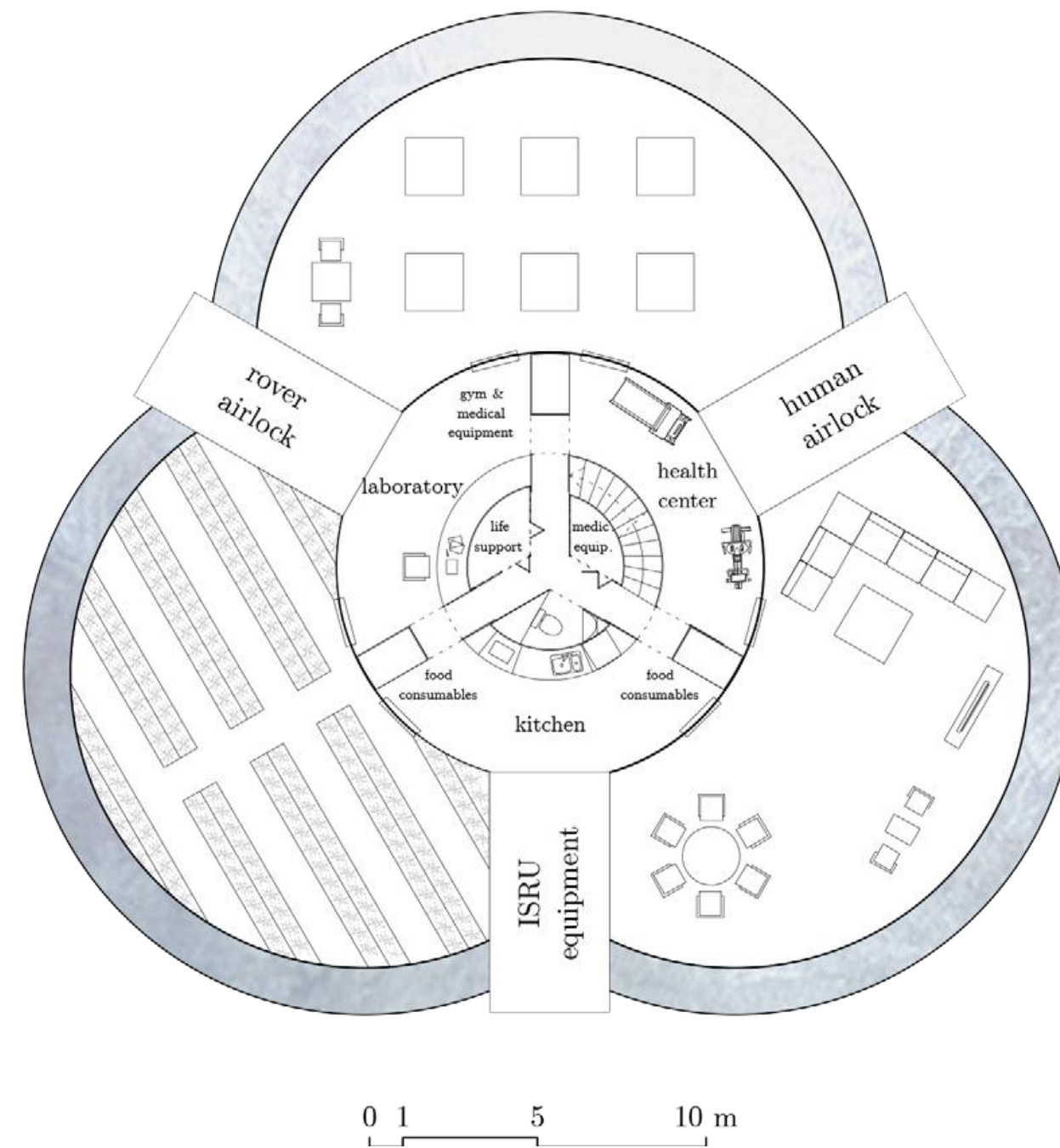


Strategy

ethylene & ice

- radiation protection
 - easy to produce
 - from polar ice
- from atmosphere
 - plastics
 - fuel

Ground floor



introduction

strategy selection

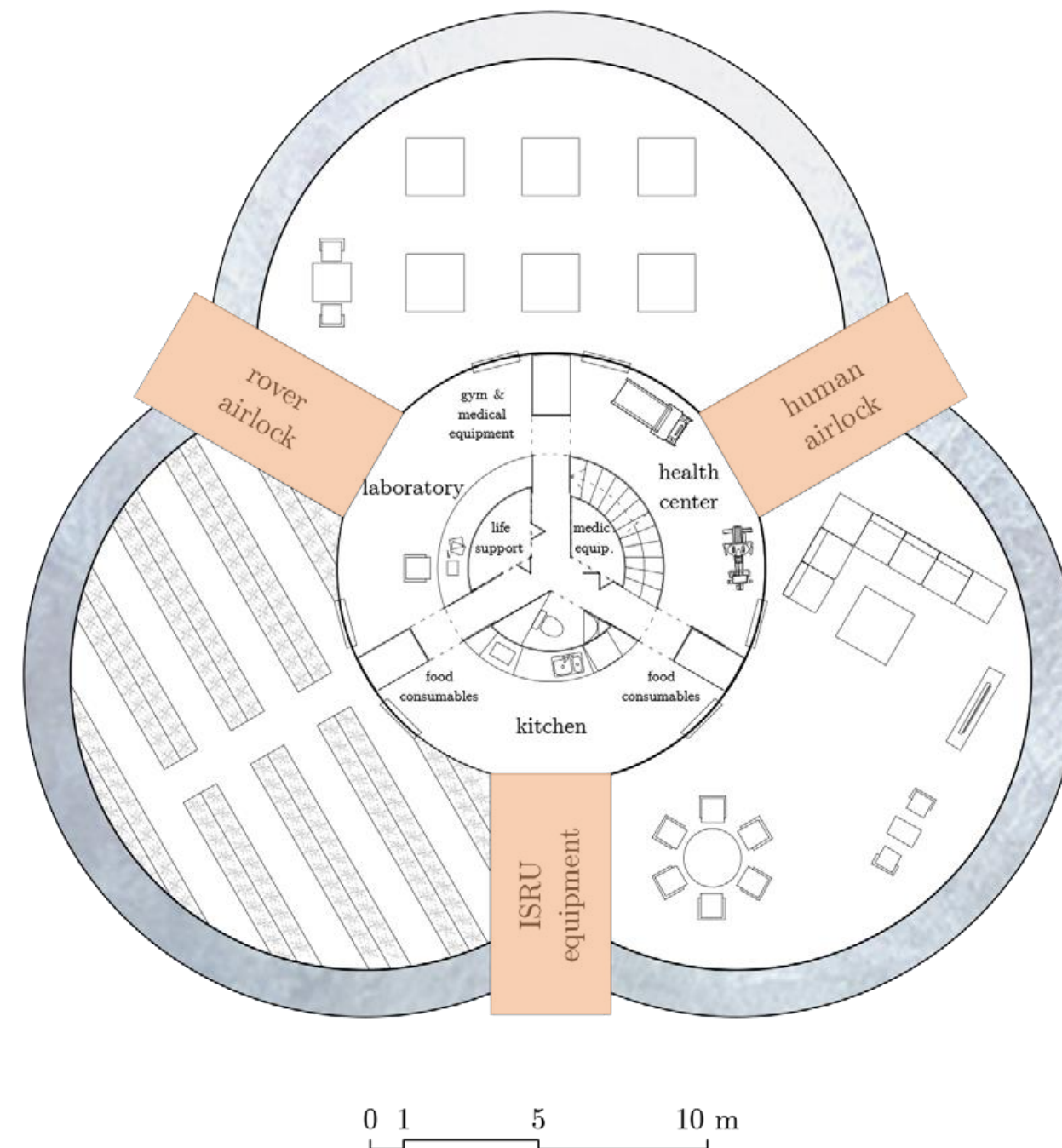
base design

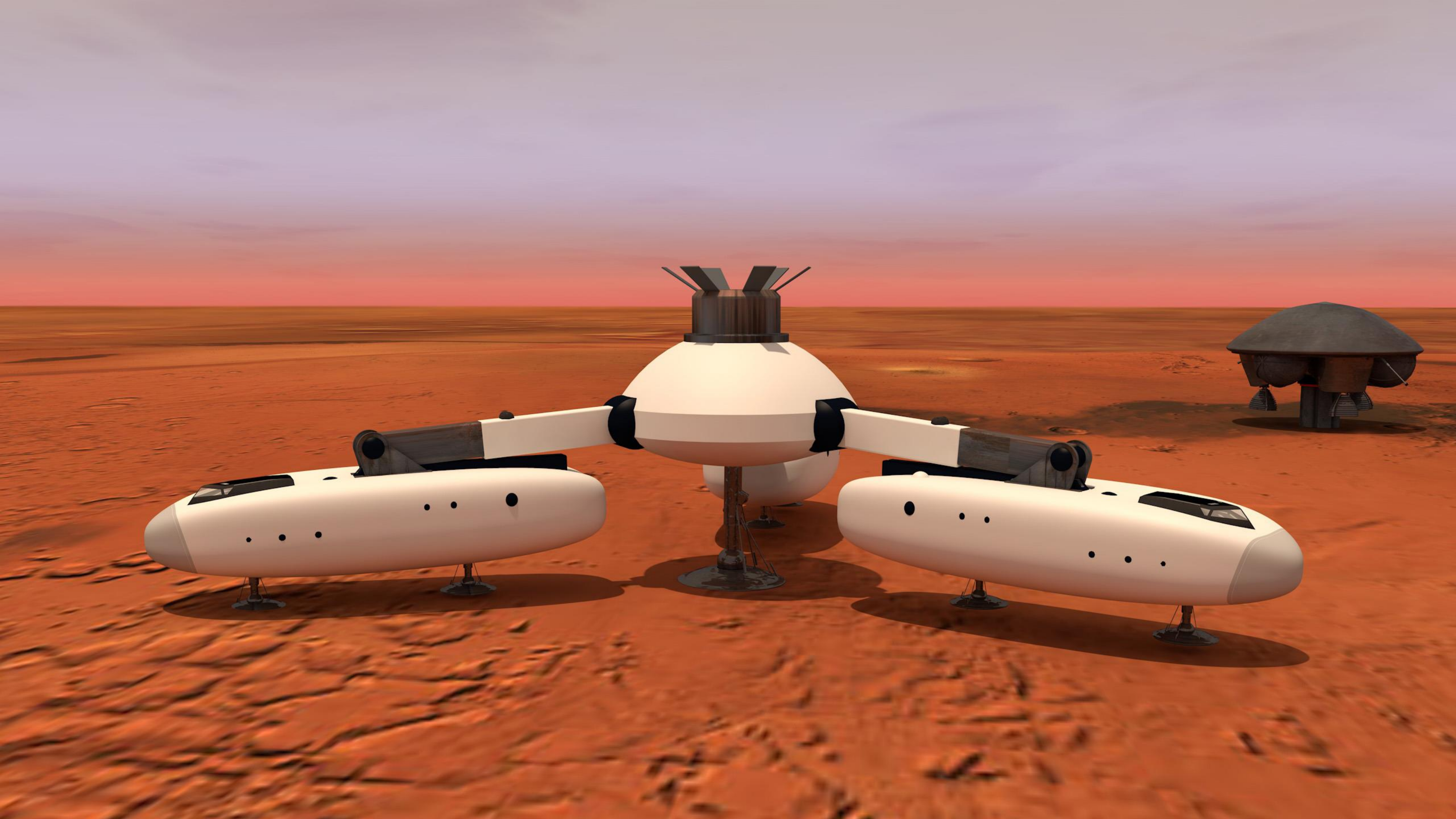
crane design

conclusions

Ground floor

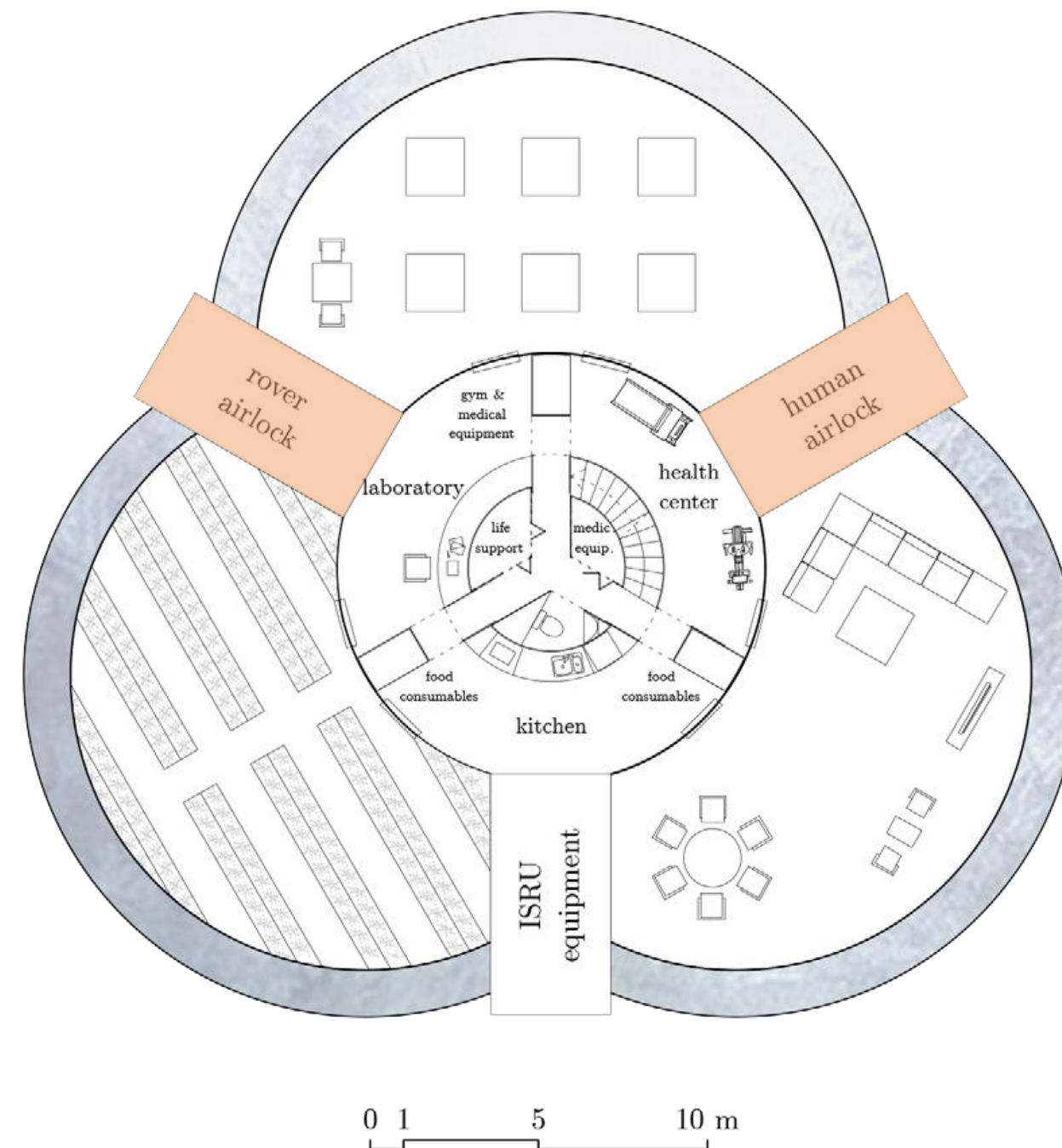
Pods



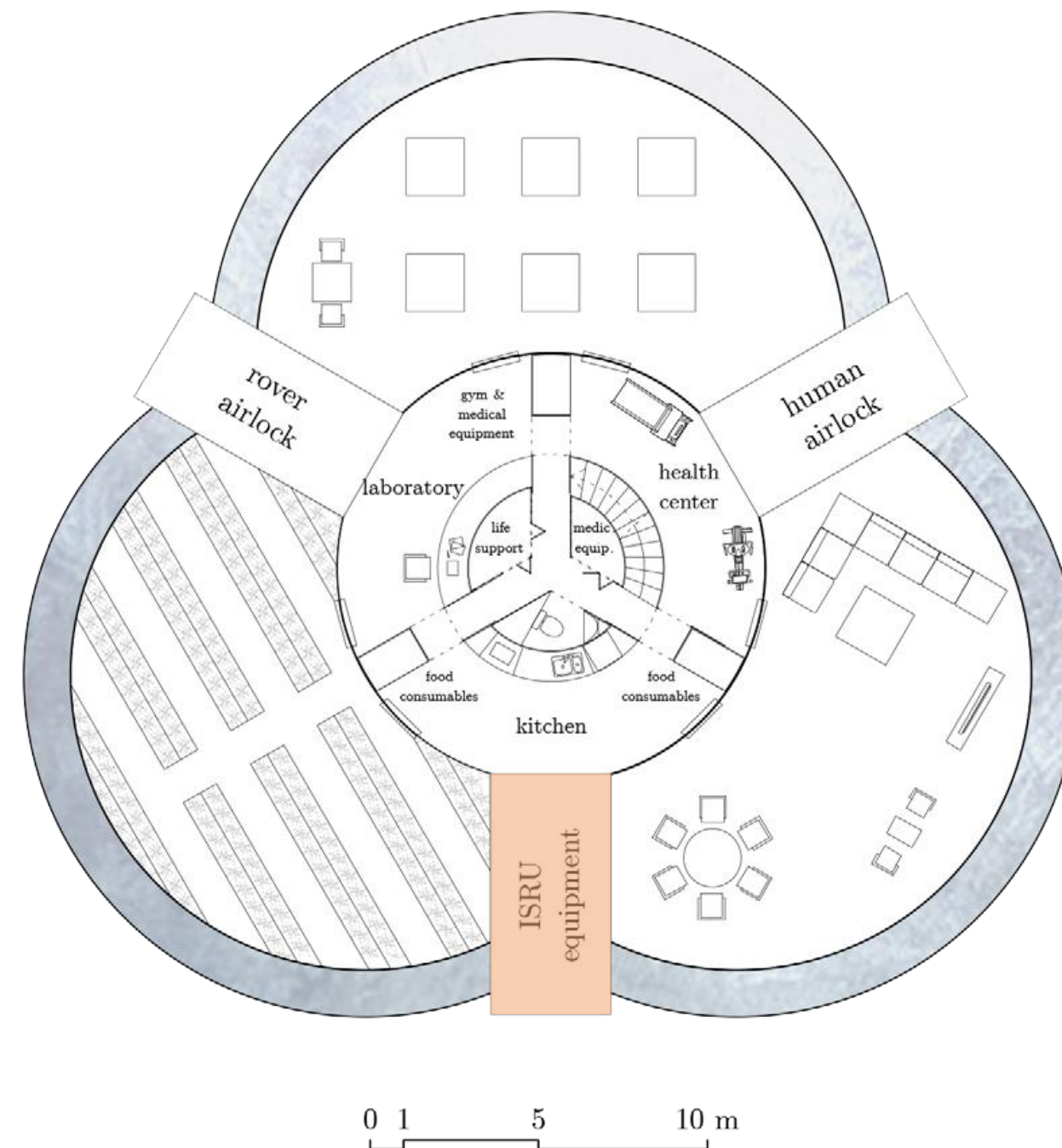


Ground floor

airlocks



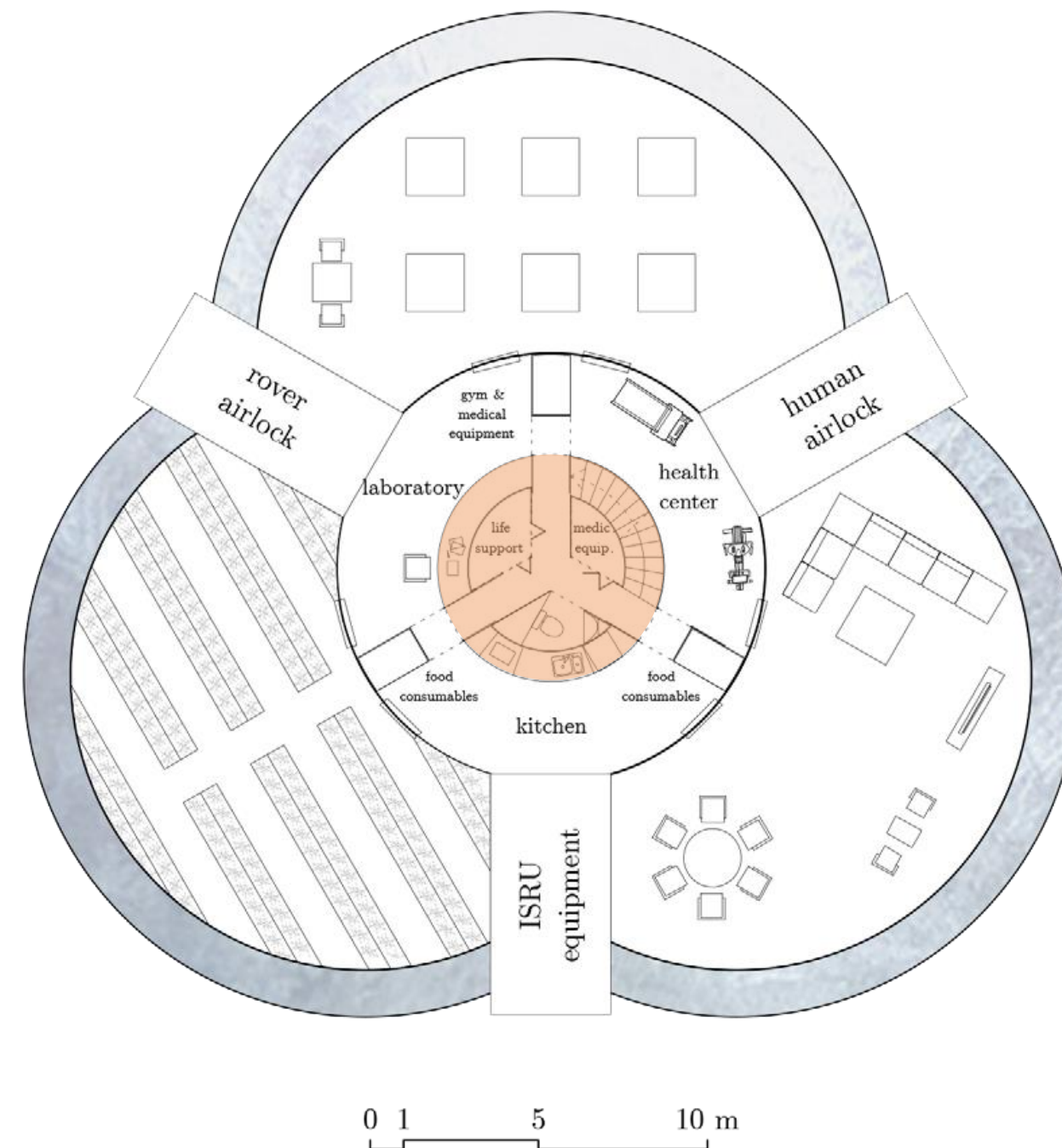
Ground floor



ISRU
equipment



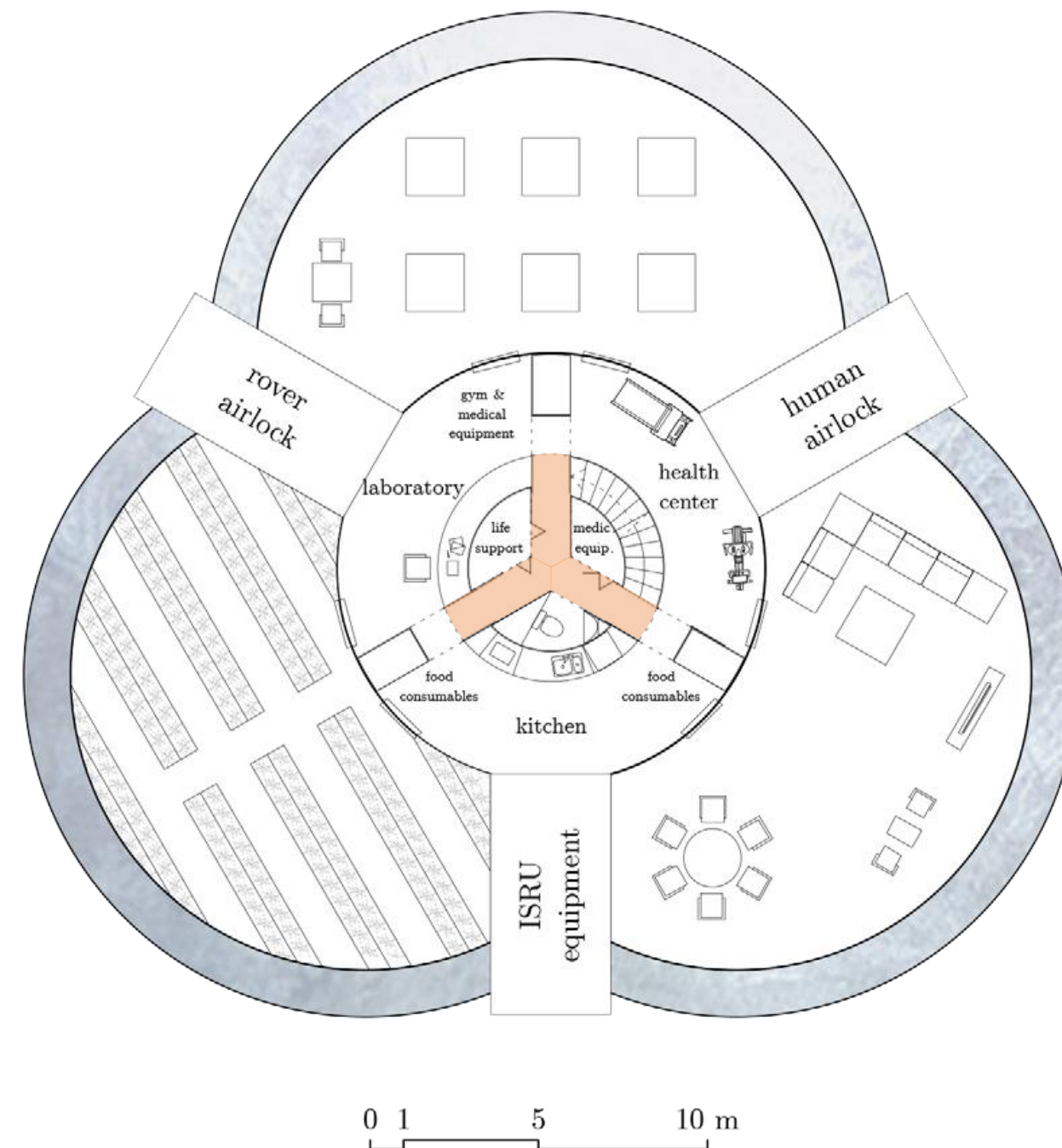
Ground floor



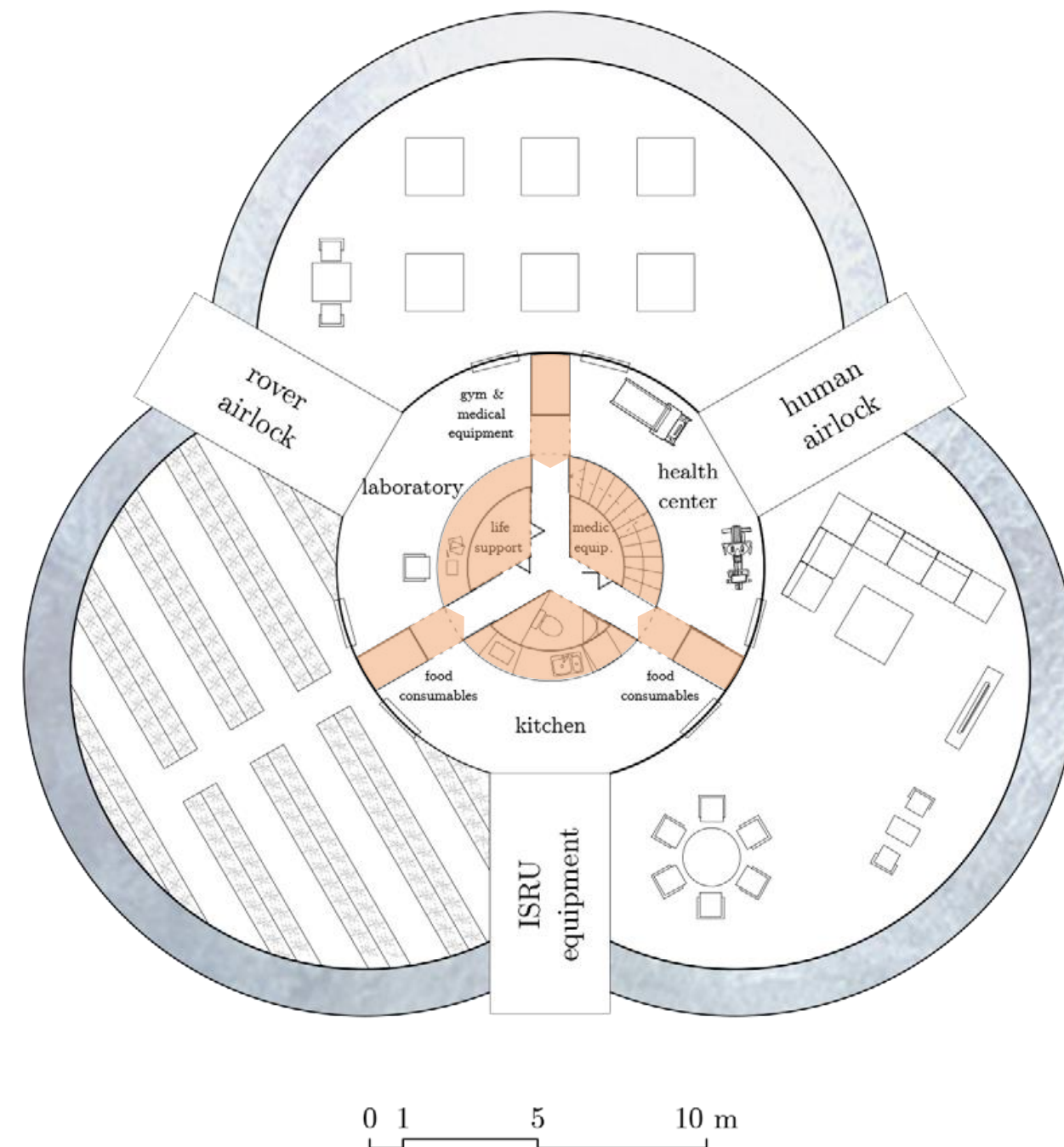
central
core

Ground floor

frames

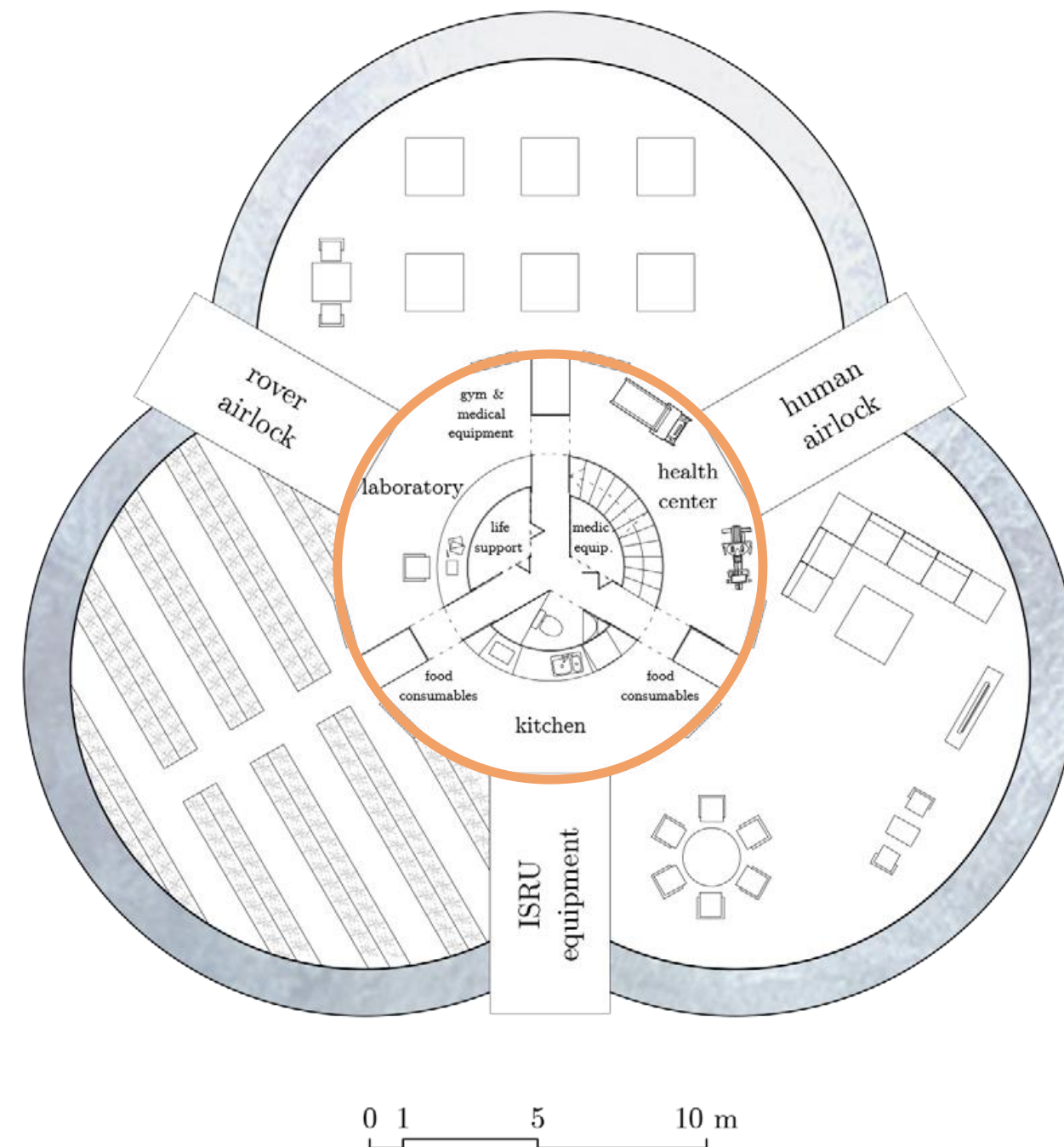


Ground floor



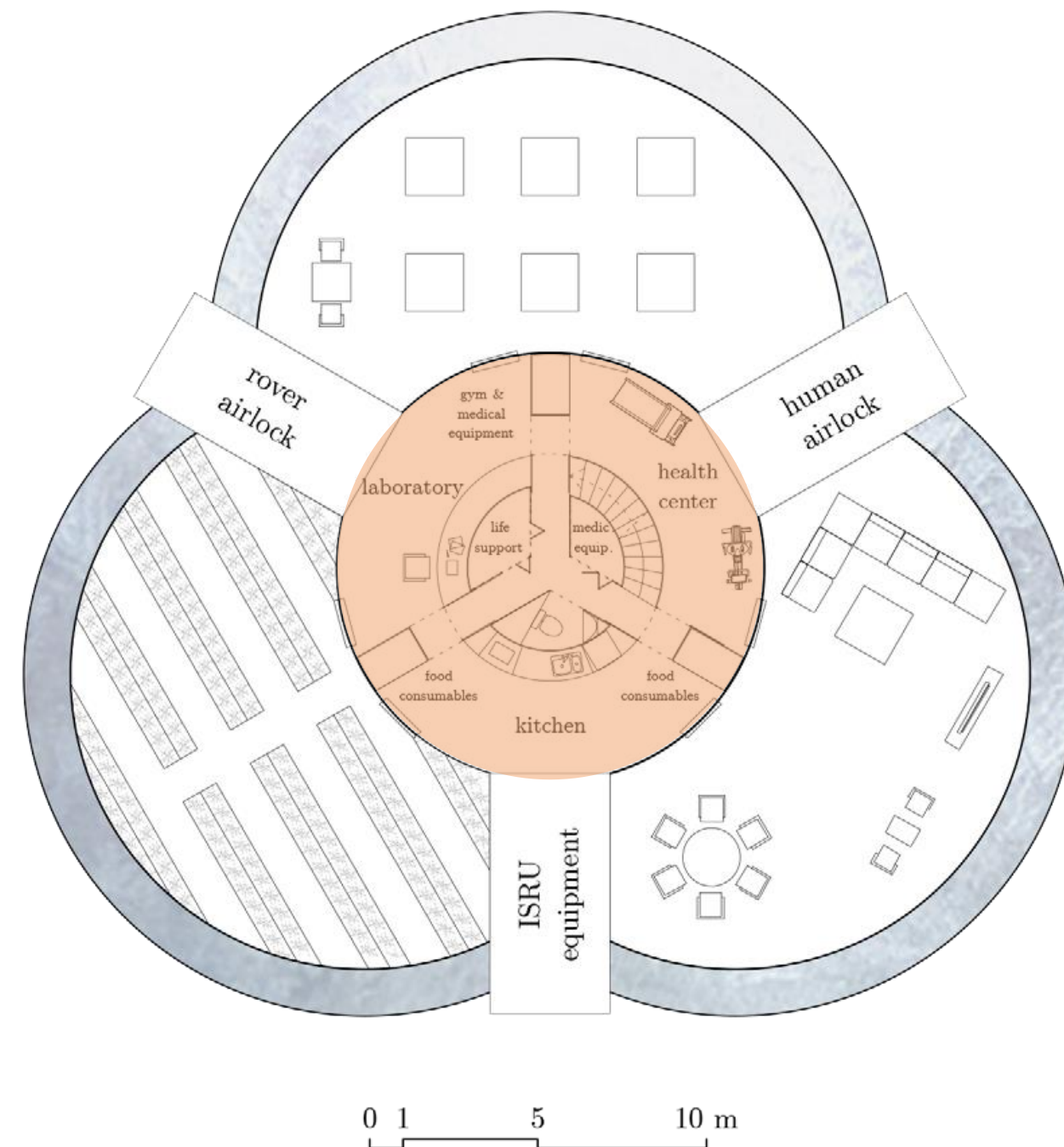
deployed
core

Ground floor



1st
membrane

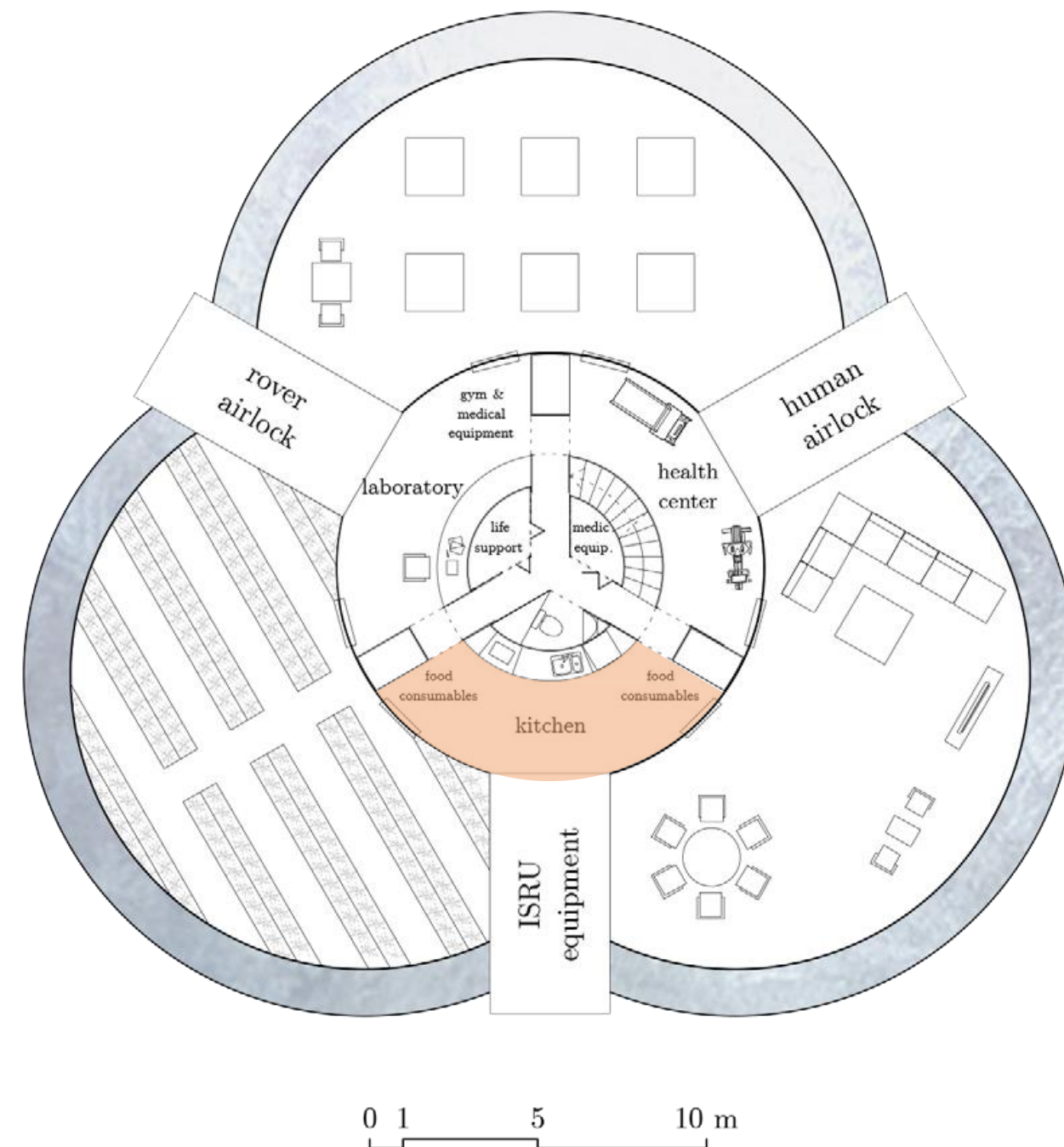
Ground floor



minimal
space

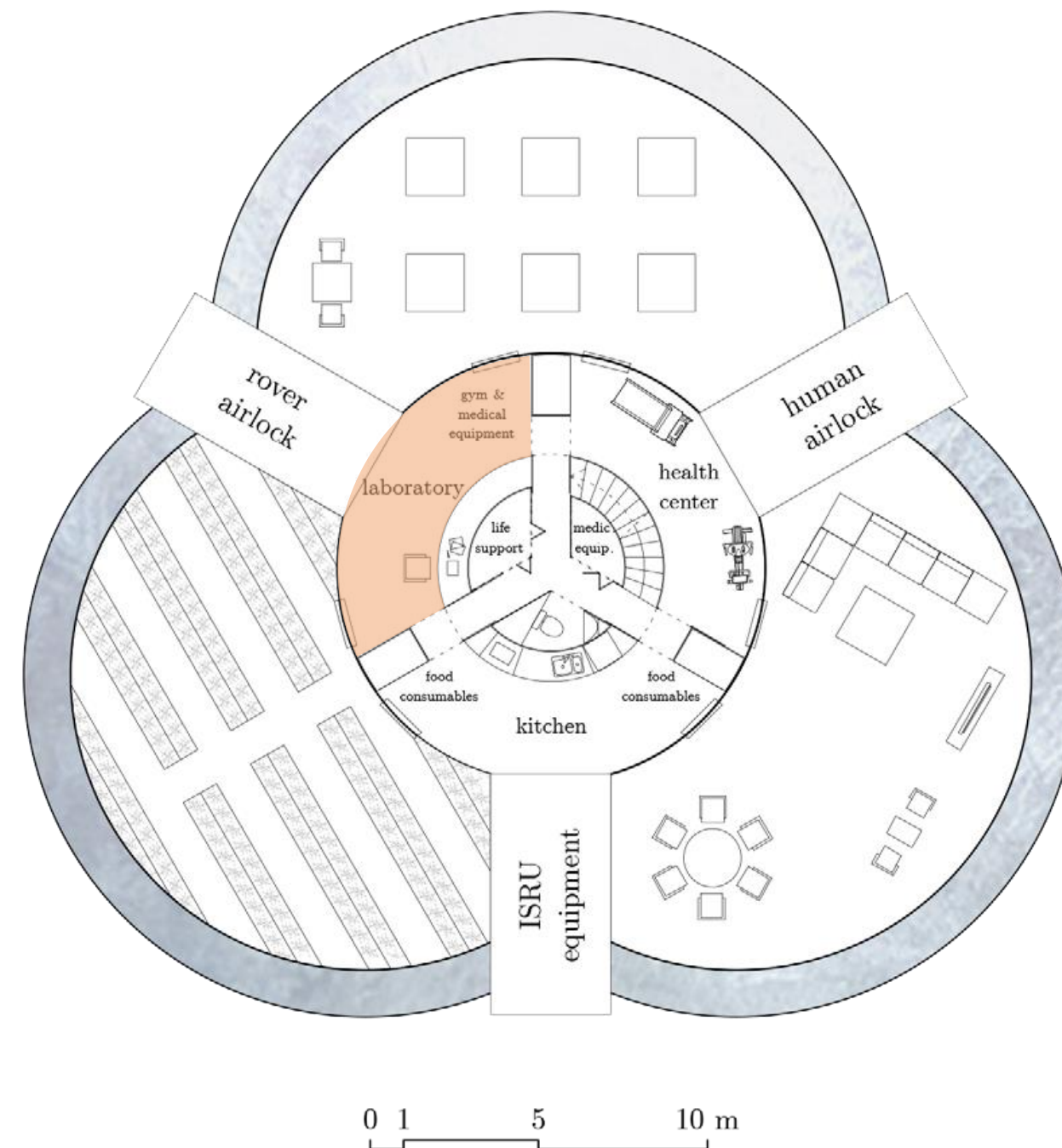
Ground floor

kitchen



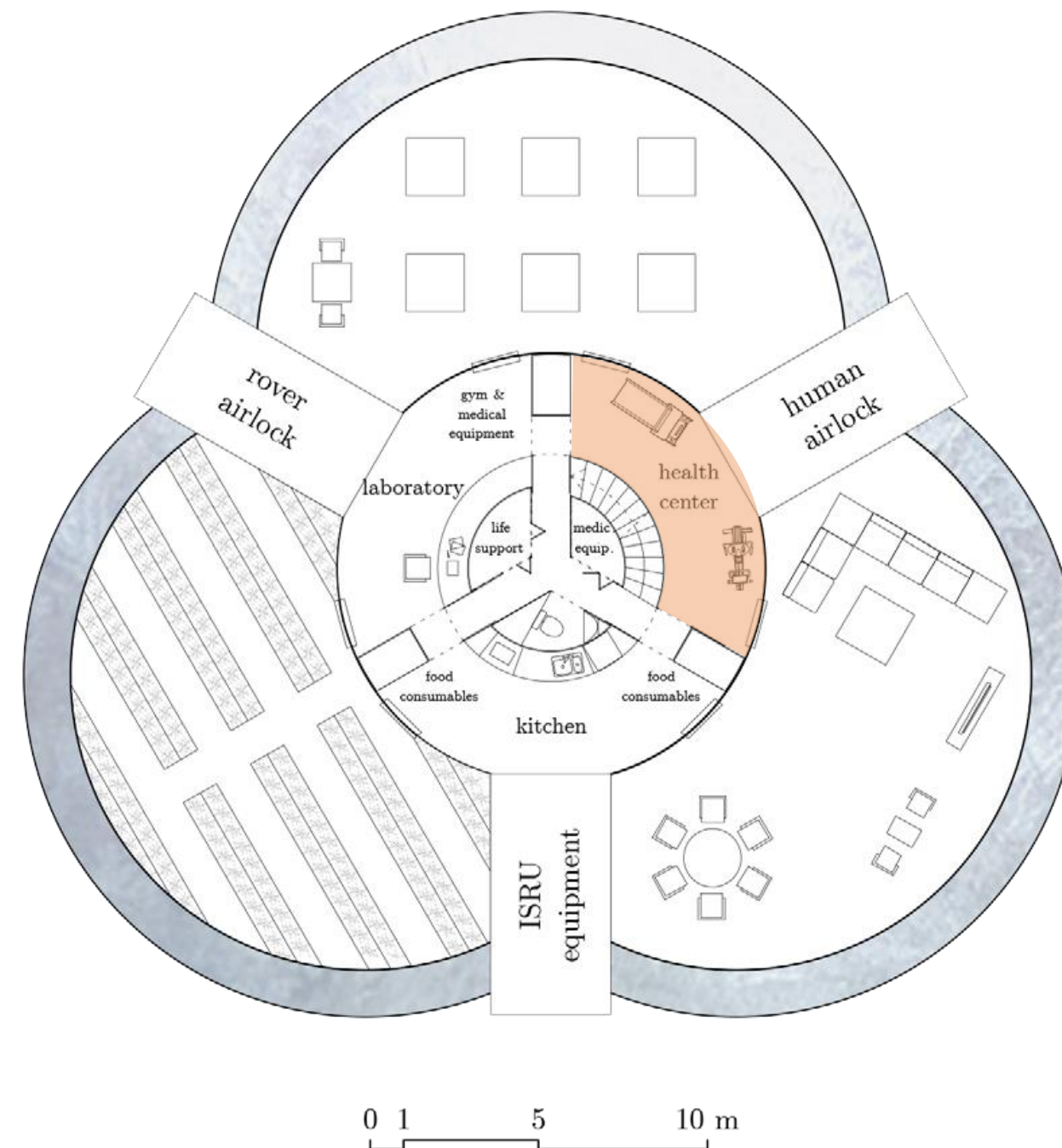
Ground floor

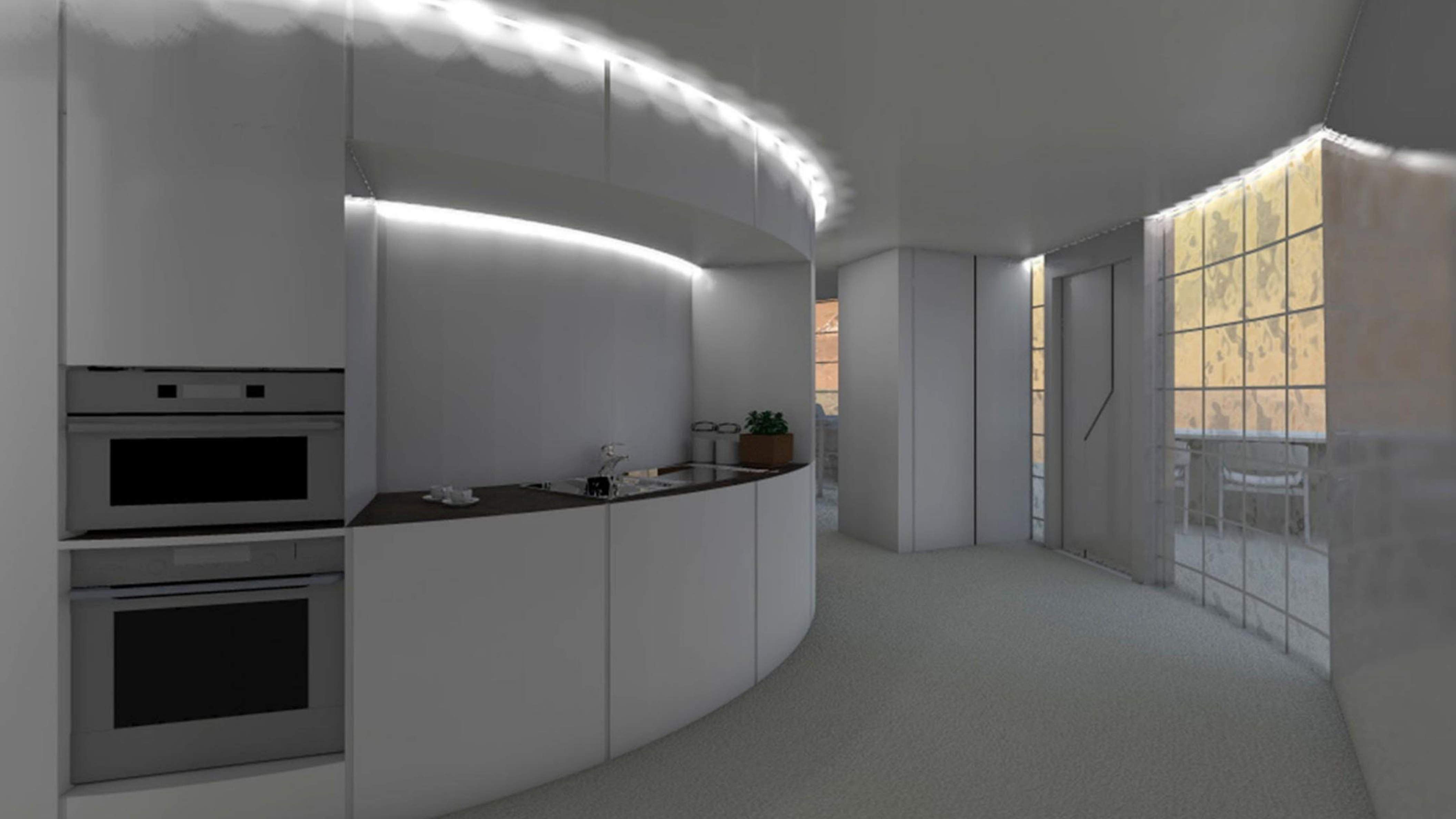
laboratory



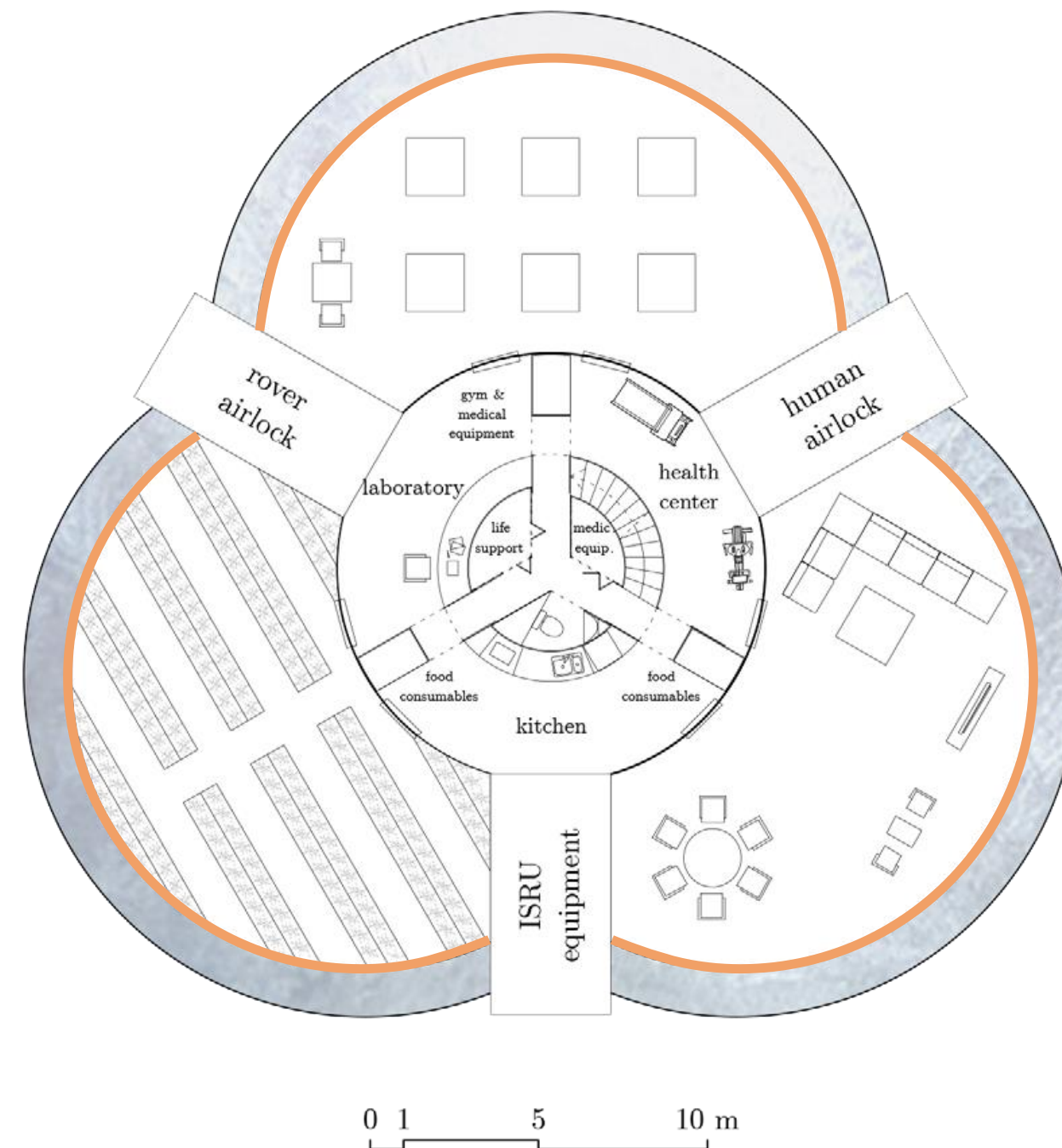
Ground floor

health
center





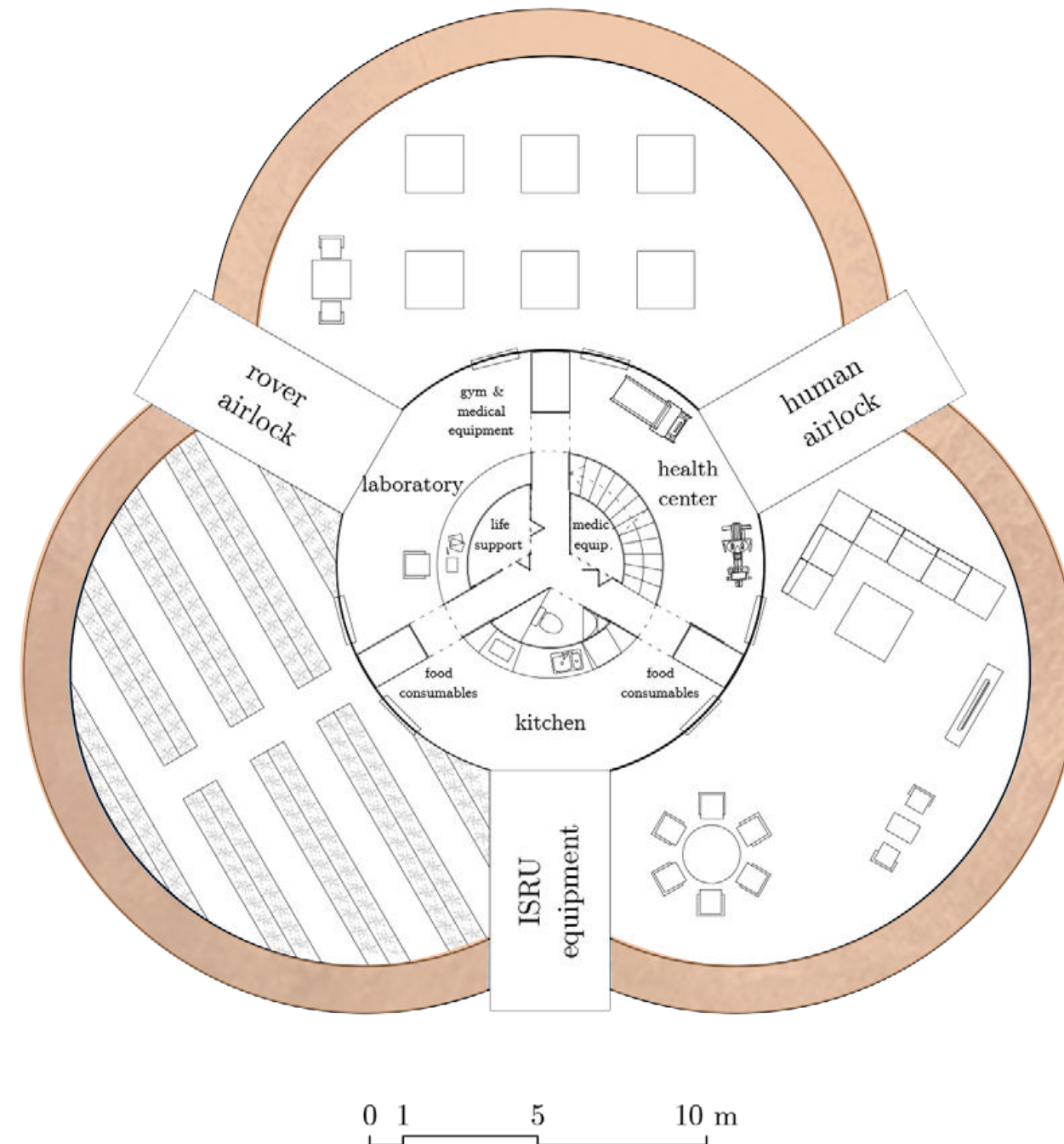
Ground floor



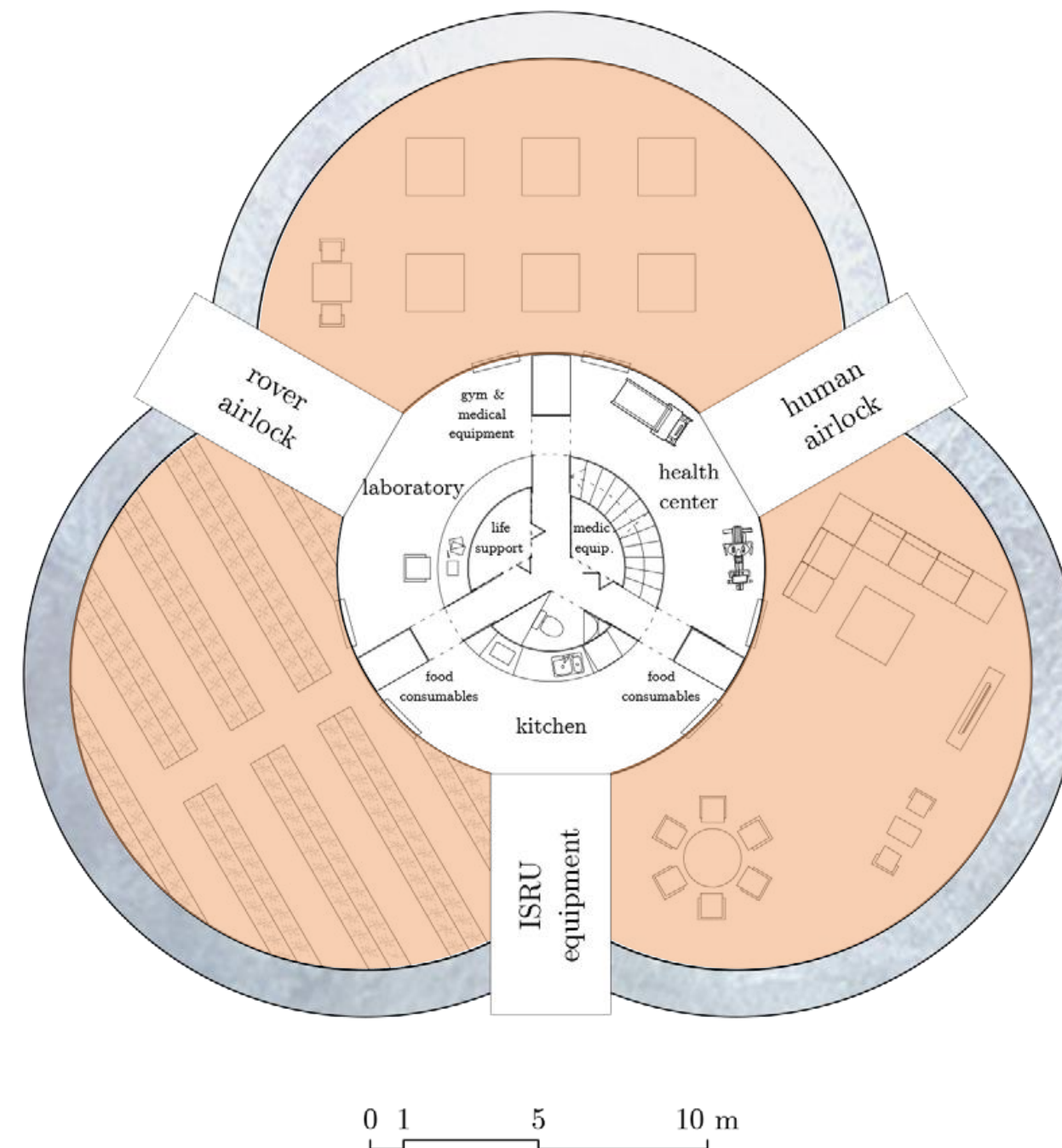
2nd
membrane

Ground floor

ice dome



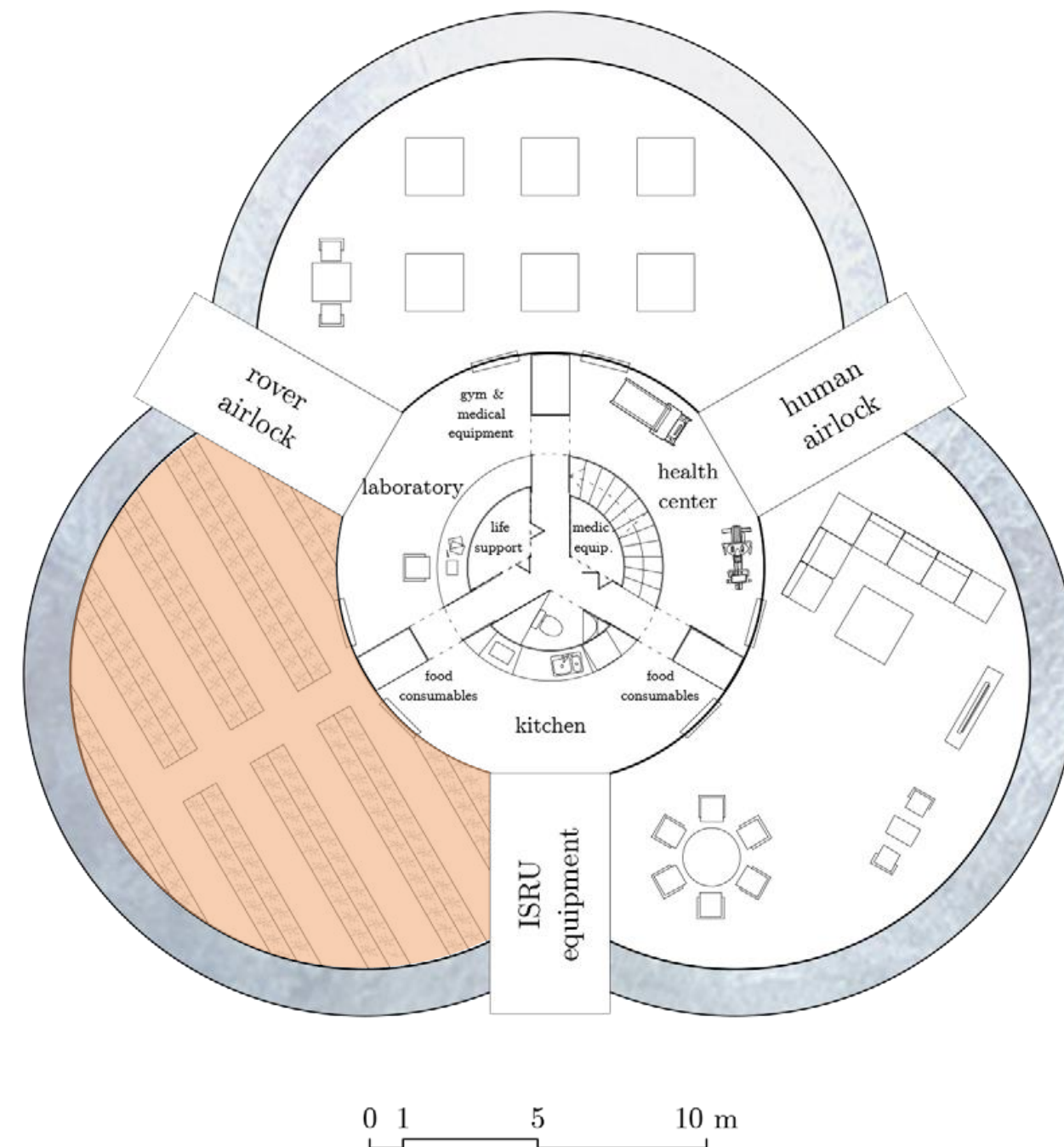
Ground floor



additional
space

Ground floor

greenhouse



introduction

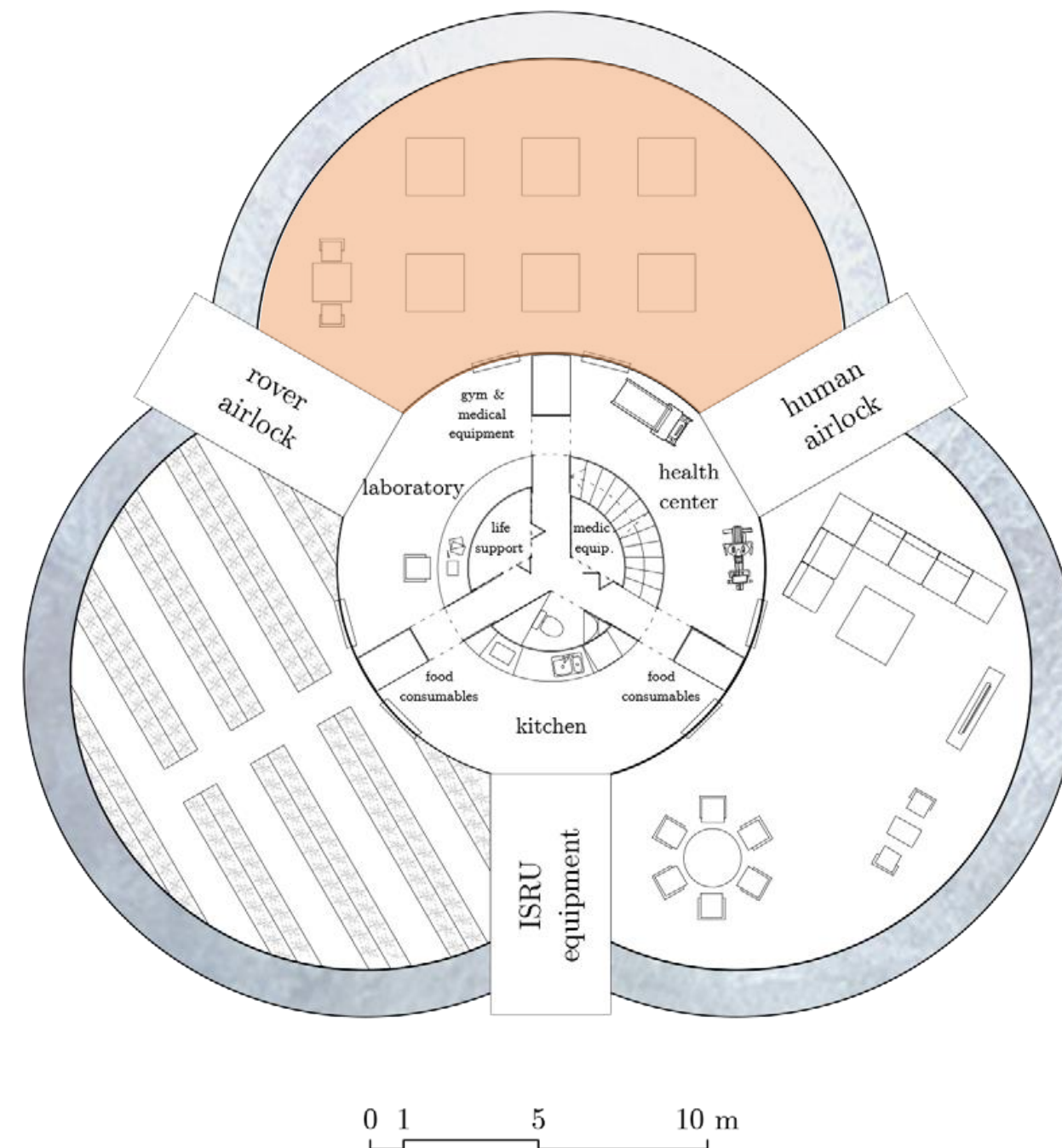
strategy selection

base design

crane design

conclusions

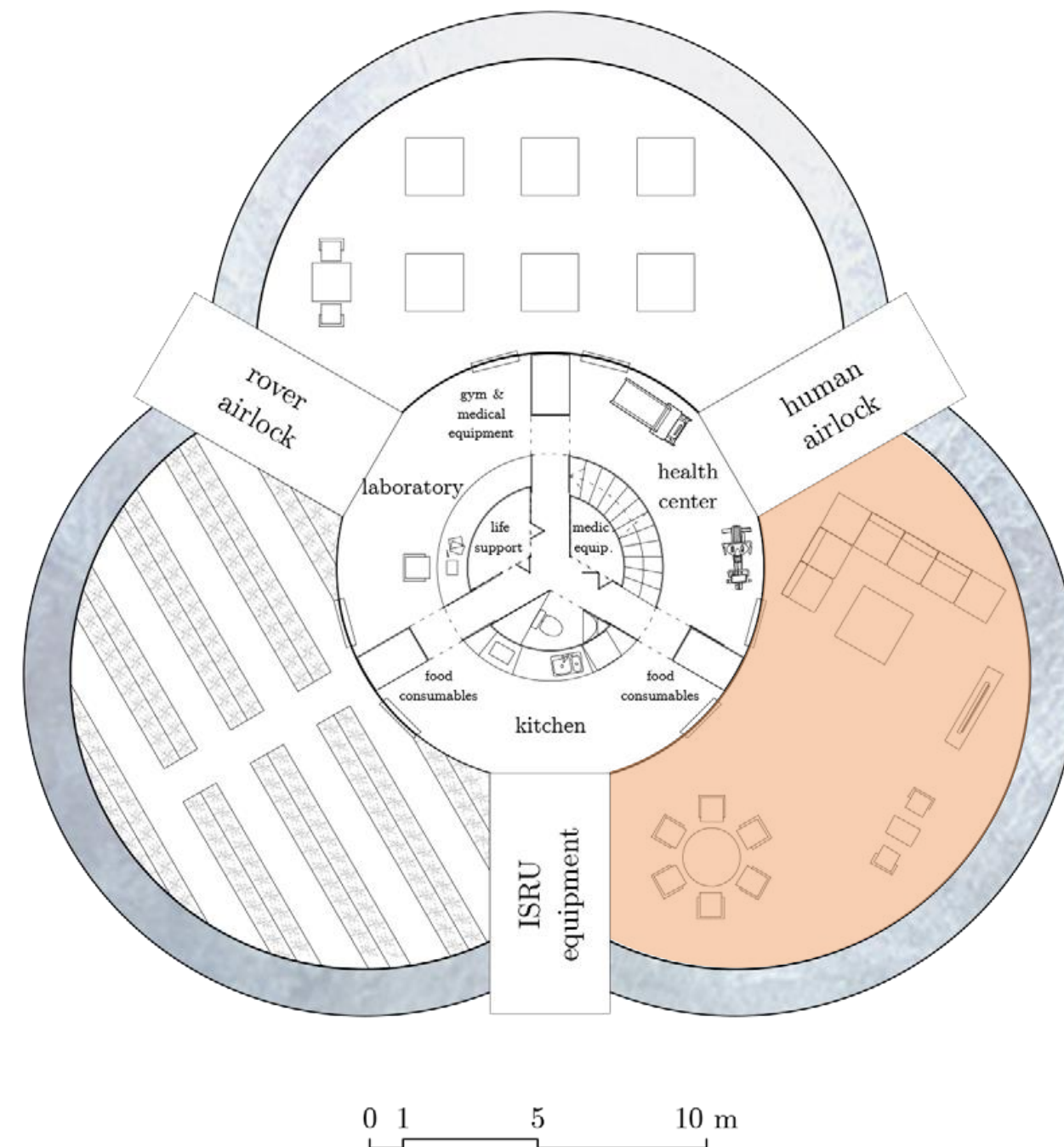
Ground floor



laboratory
extension

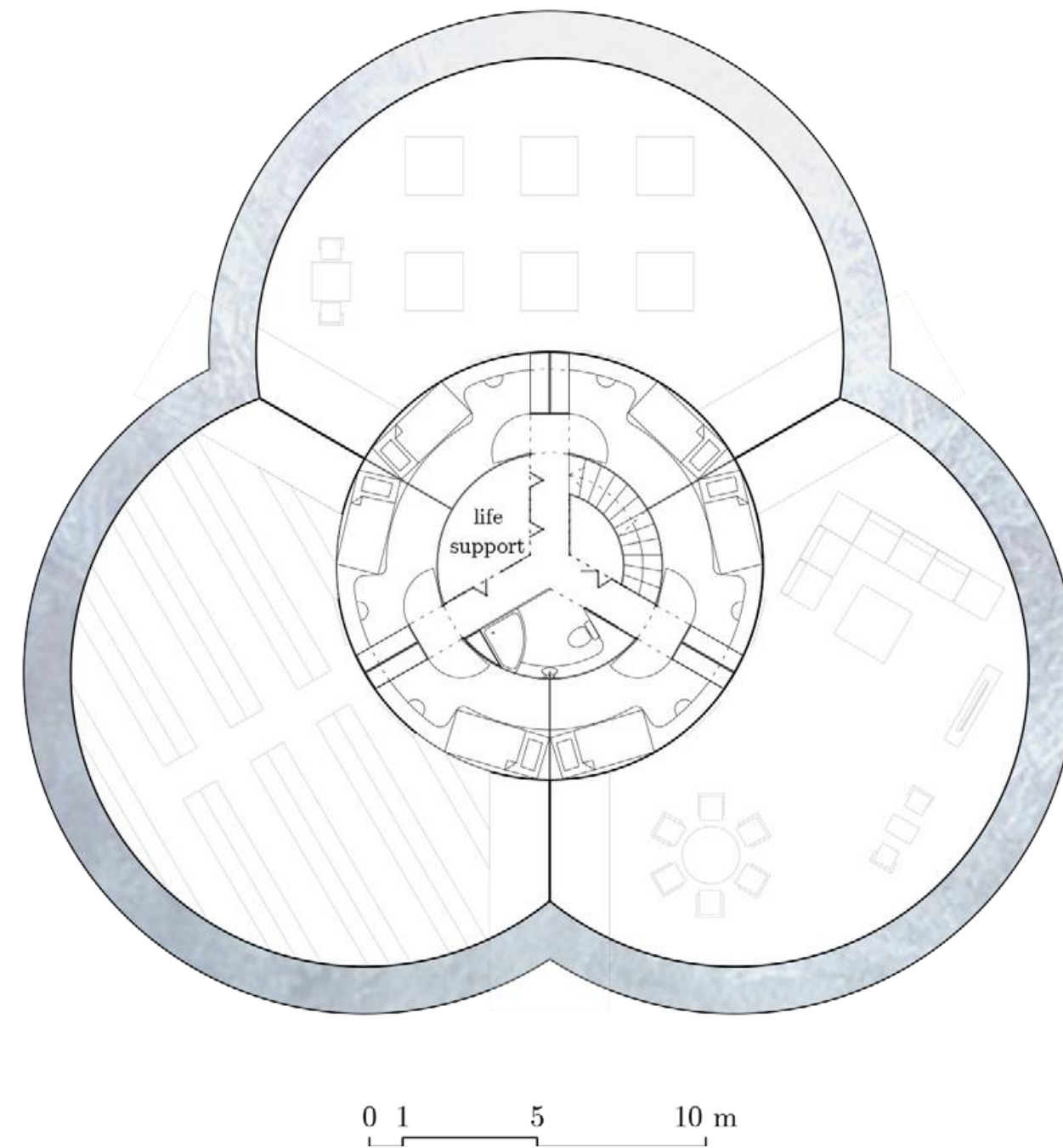
Ground floor

living





First Floor



introduction

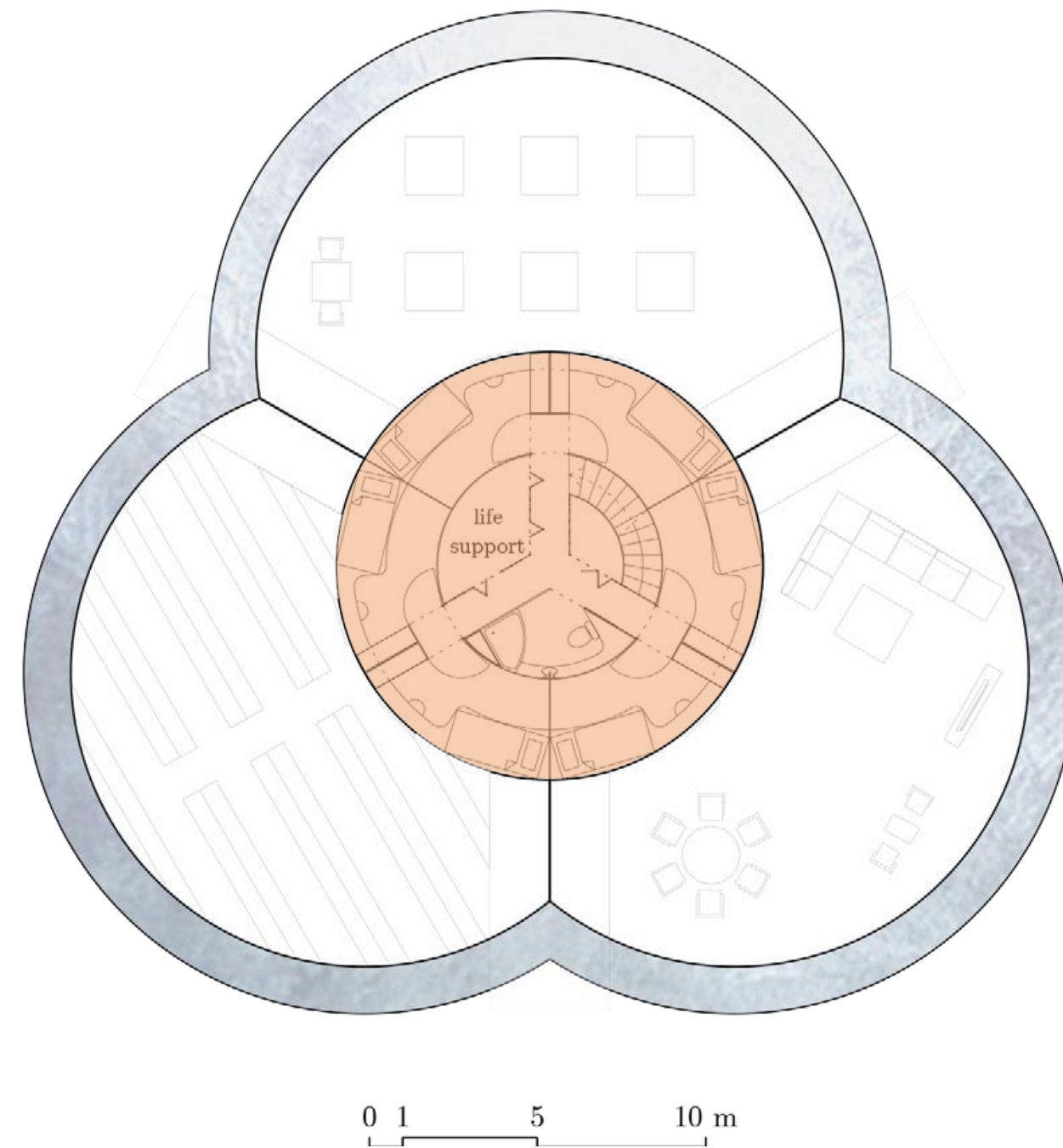
strategy selection

base design

crane design

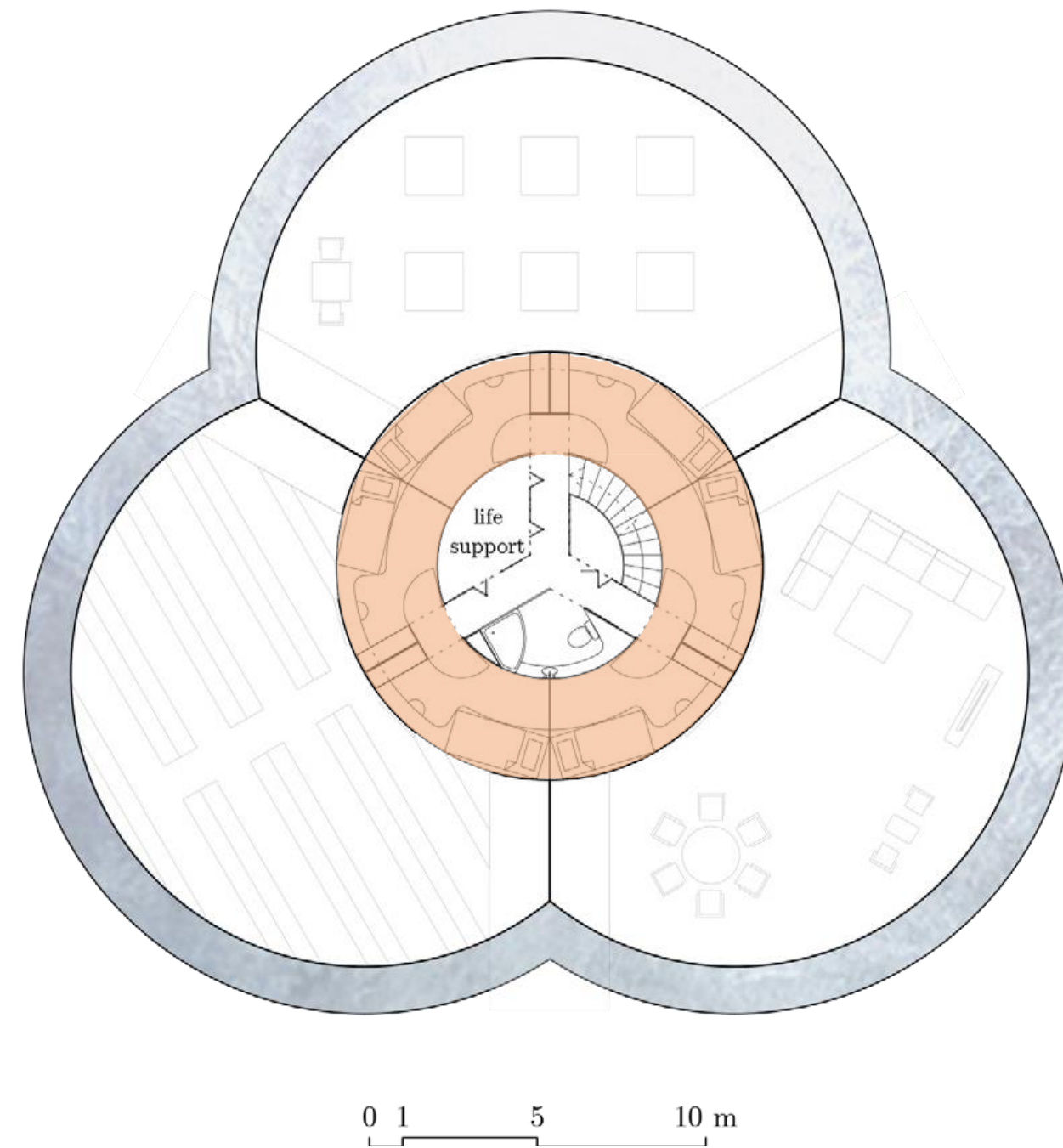
conclusions

First Floor



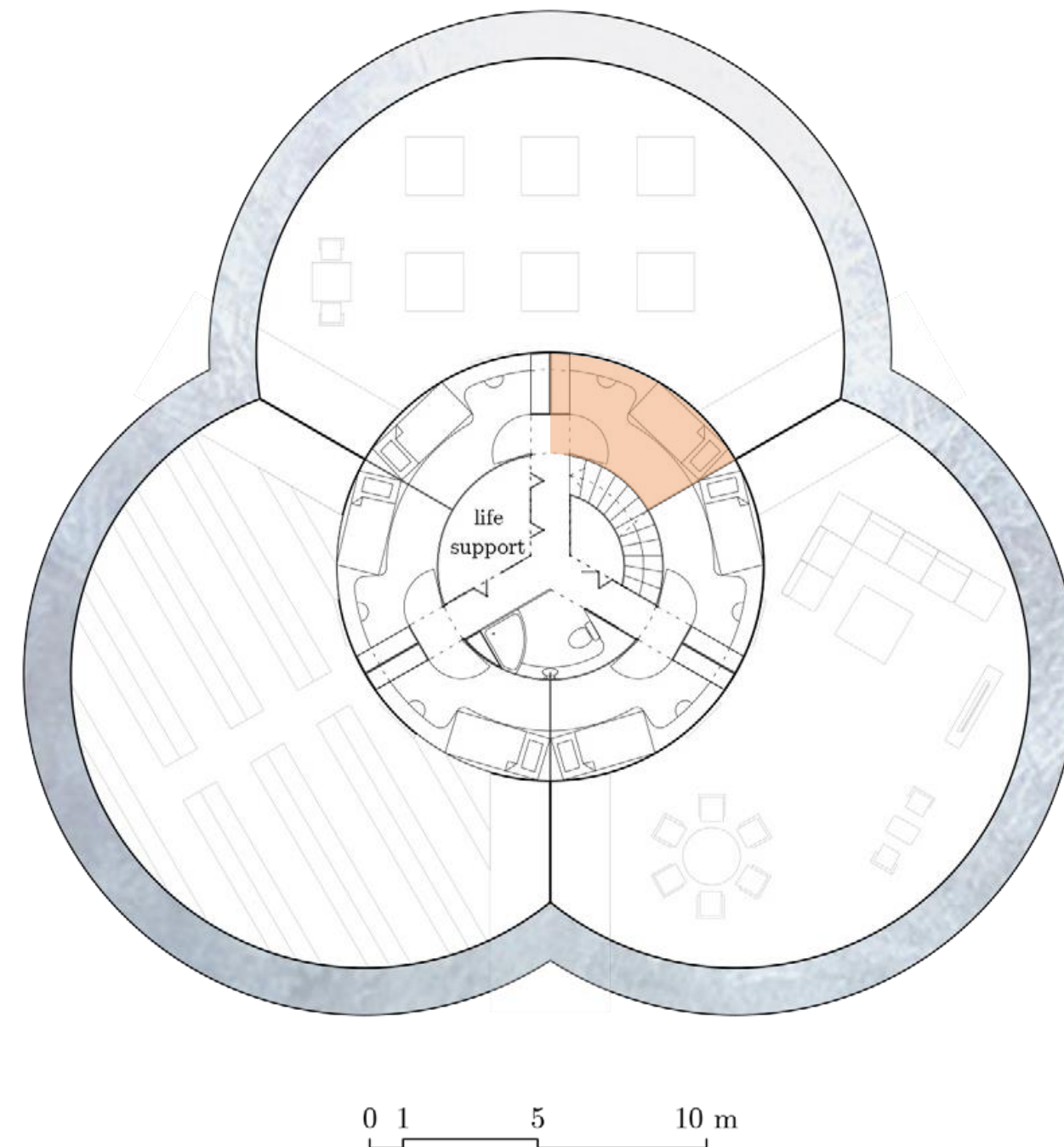
central
core

First Floor



crew
quarters

First Floor

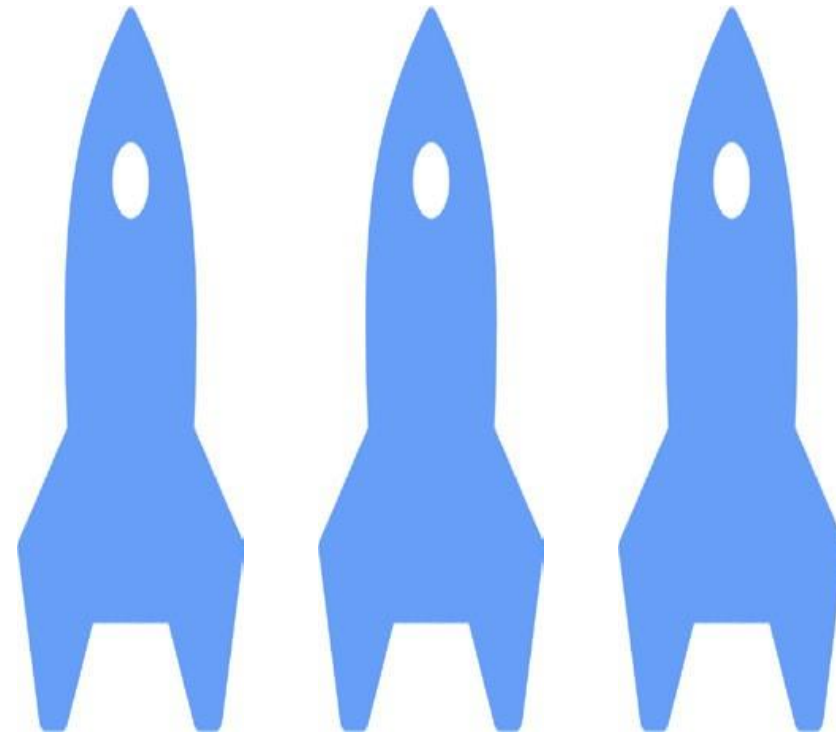


individual
crew
quarter

Crewed Mars Missions

Phase	Mass
Phase 1 - base	51,390 kg
Phase 2 - crew	57,685 kg
Total	109,075 kg

Crewed Mars Missions



3

Mars orbit-to-surface
descends minimum

Robotic Mars Missions



SkyCrane for Curiosity
image courtesy of
NASA/JPL-Caltech

introduction

strategy selection

base design

crane design

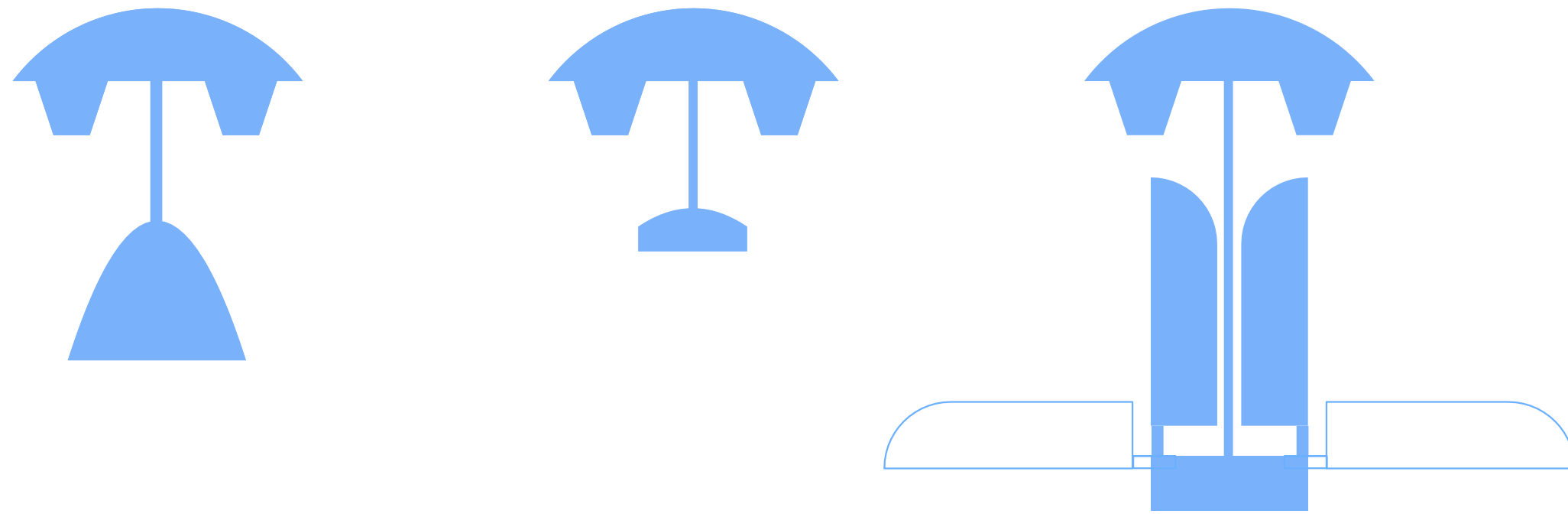
conclusions



Features

- 1 Modular capacity
- 2 Reusable
- 3 Refuellable

Modular Capacity



Launch Escape System
image courtesy of NASA

introduction

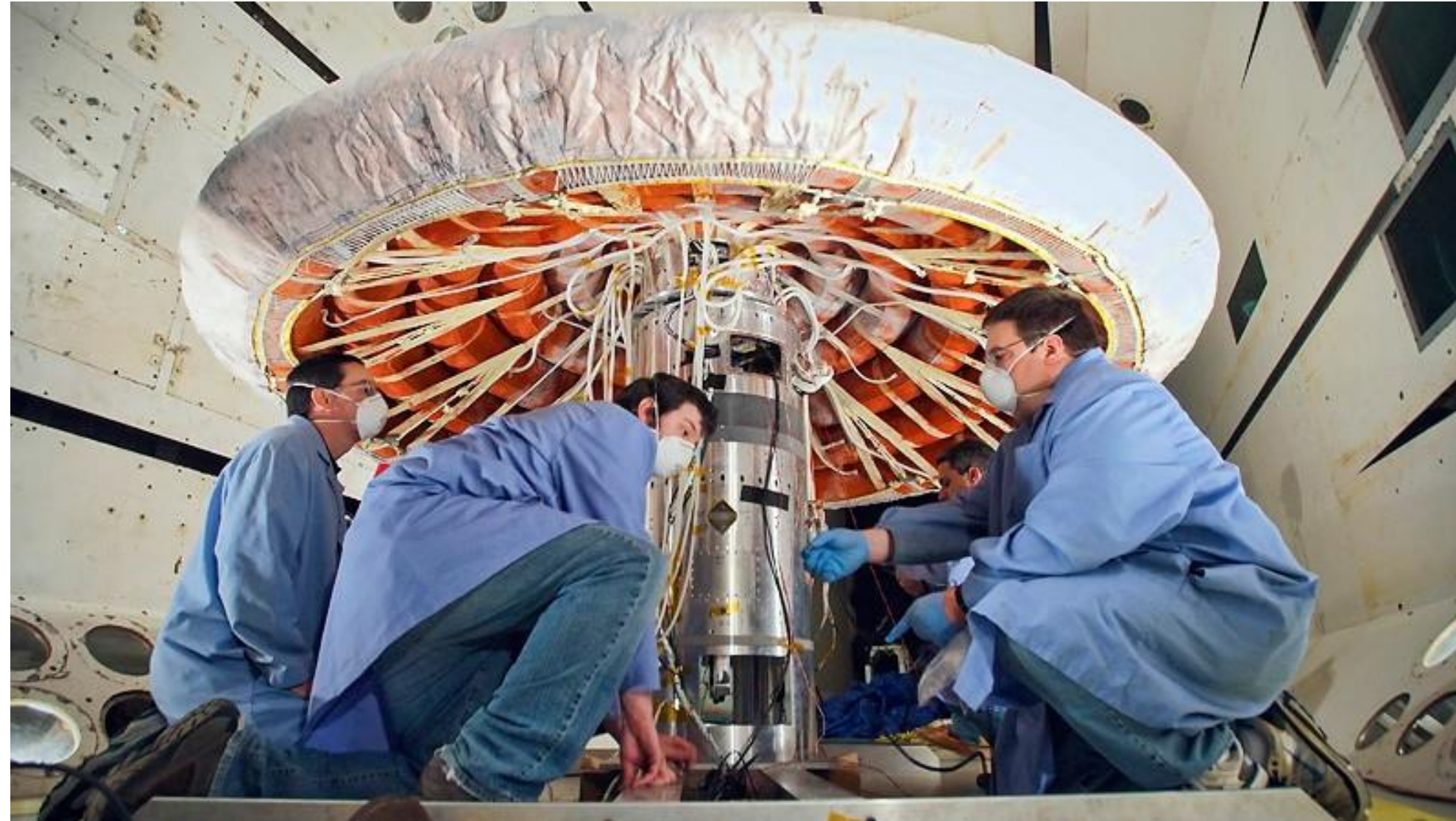
strategy selection

base design

crane design

conclusions

Reusable



IRVE-3
image courtesy of NASA

introduction

strategy selection

base design

crane design

conclusions

Refuellable

surface

base

orbit

introduction

strategy selection

base design

crane design

conclusions

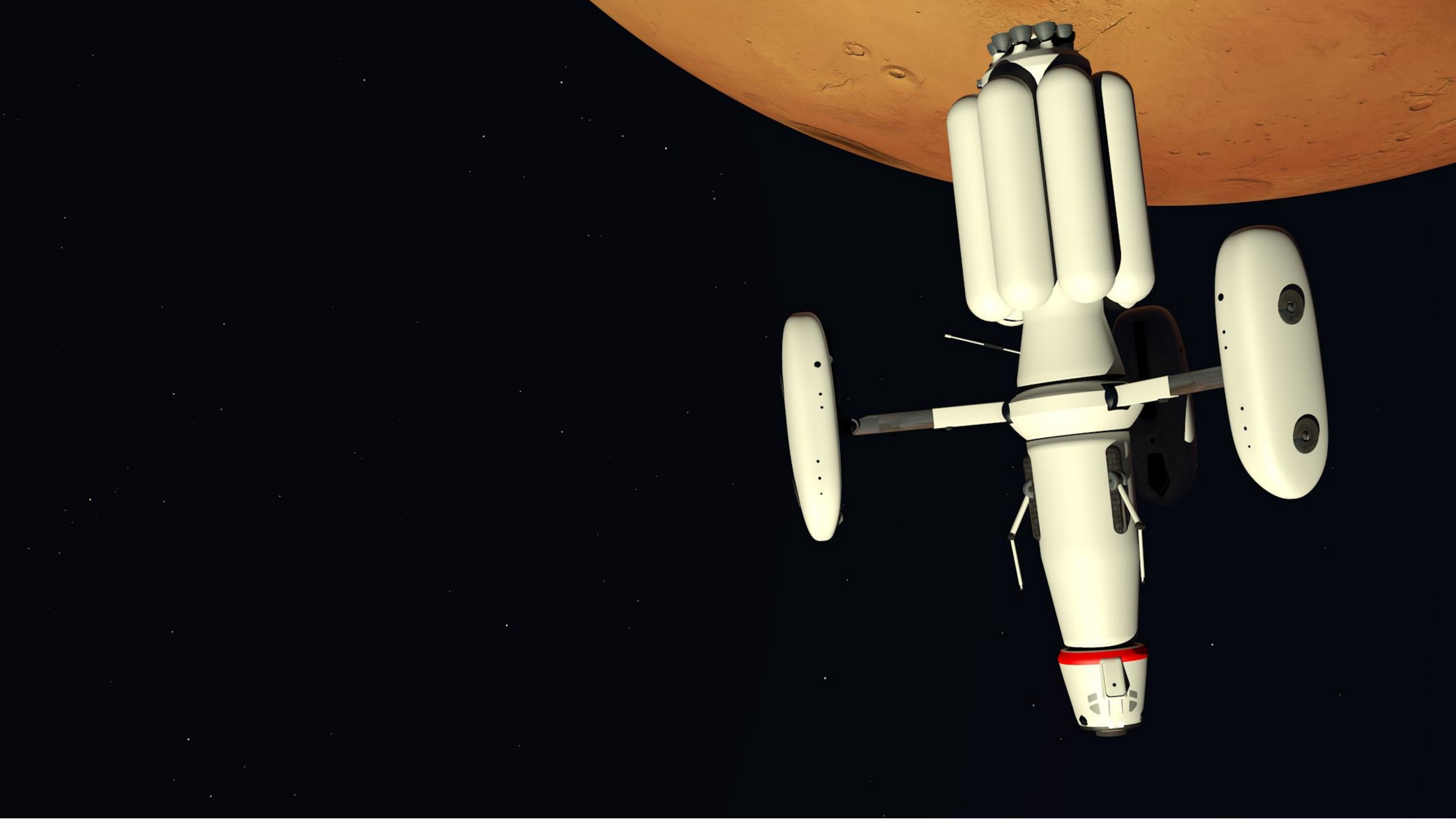
Launch Capacity to Mars

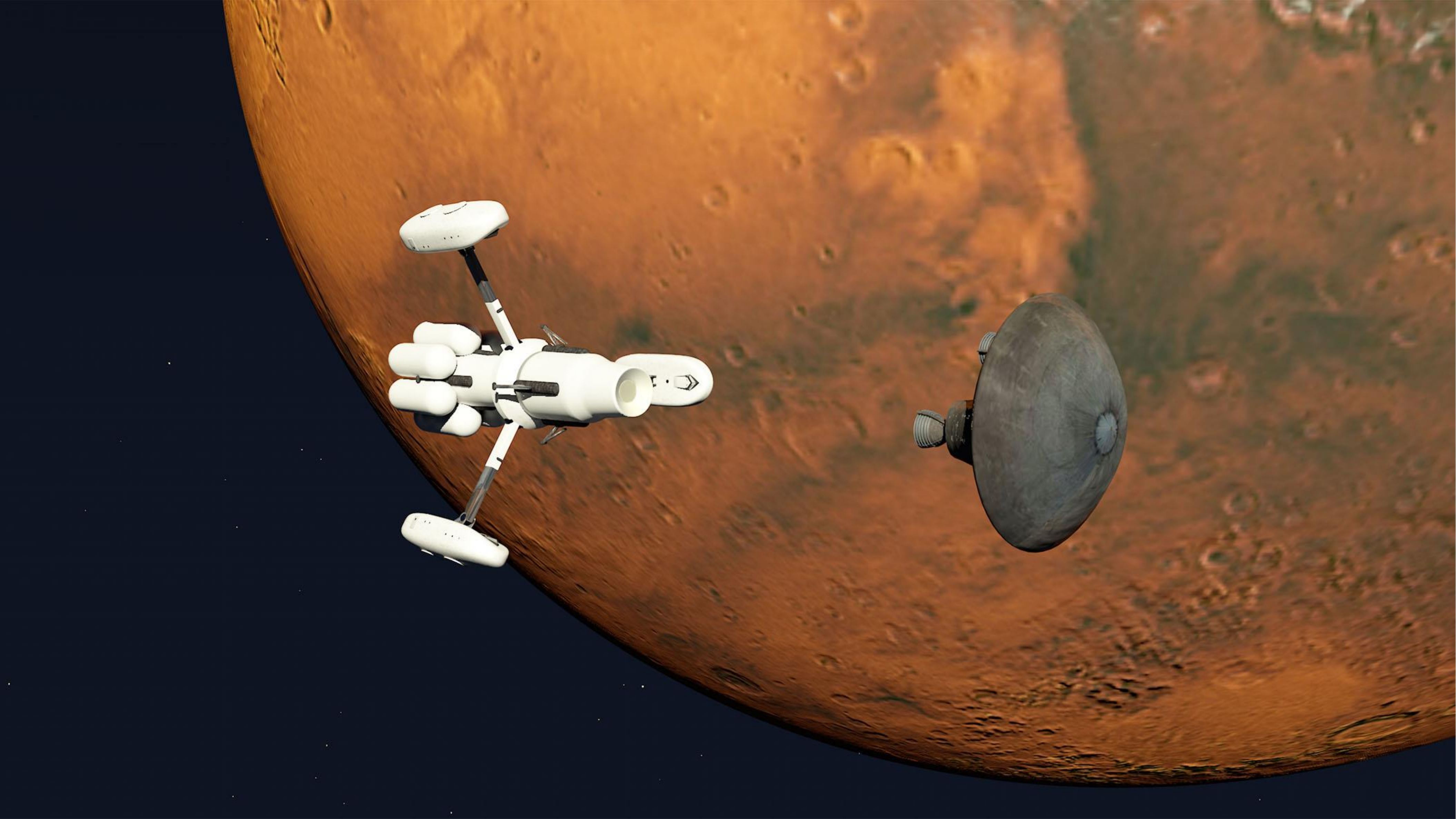


NASA
Space Launch
System

31-35
tons

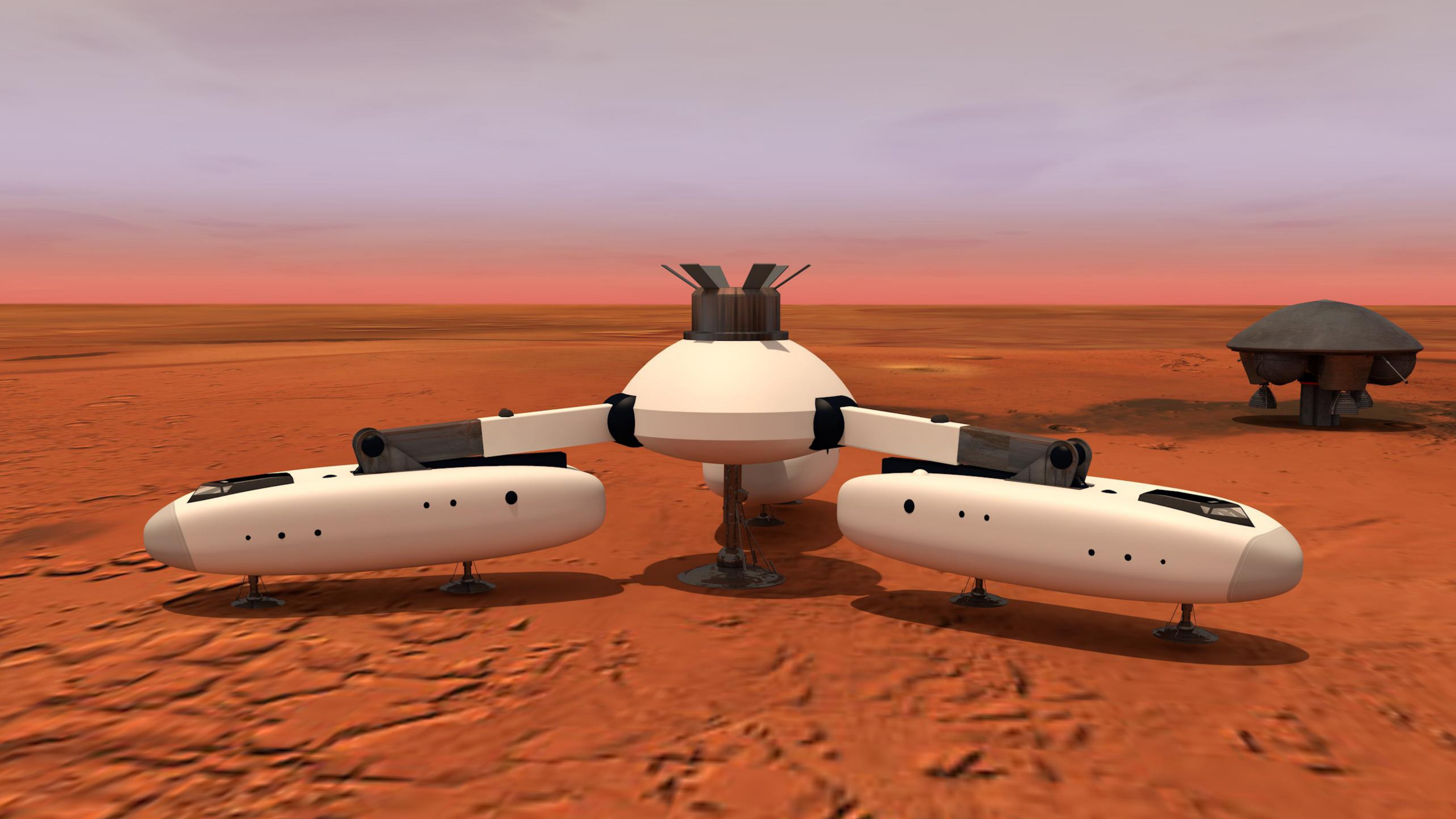
image courtesy of NASA/MSFC
image courtesy of SpaceX





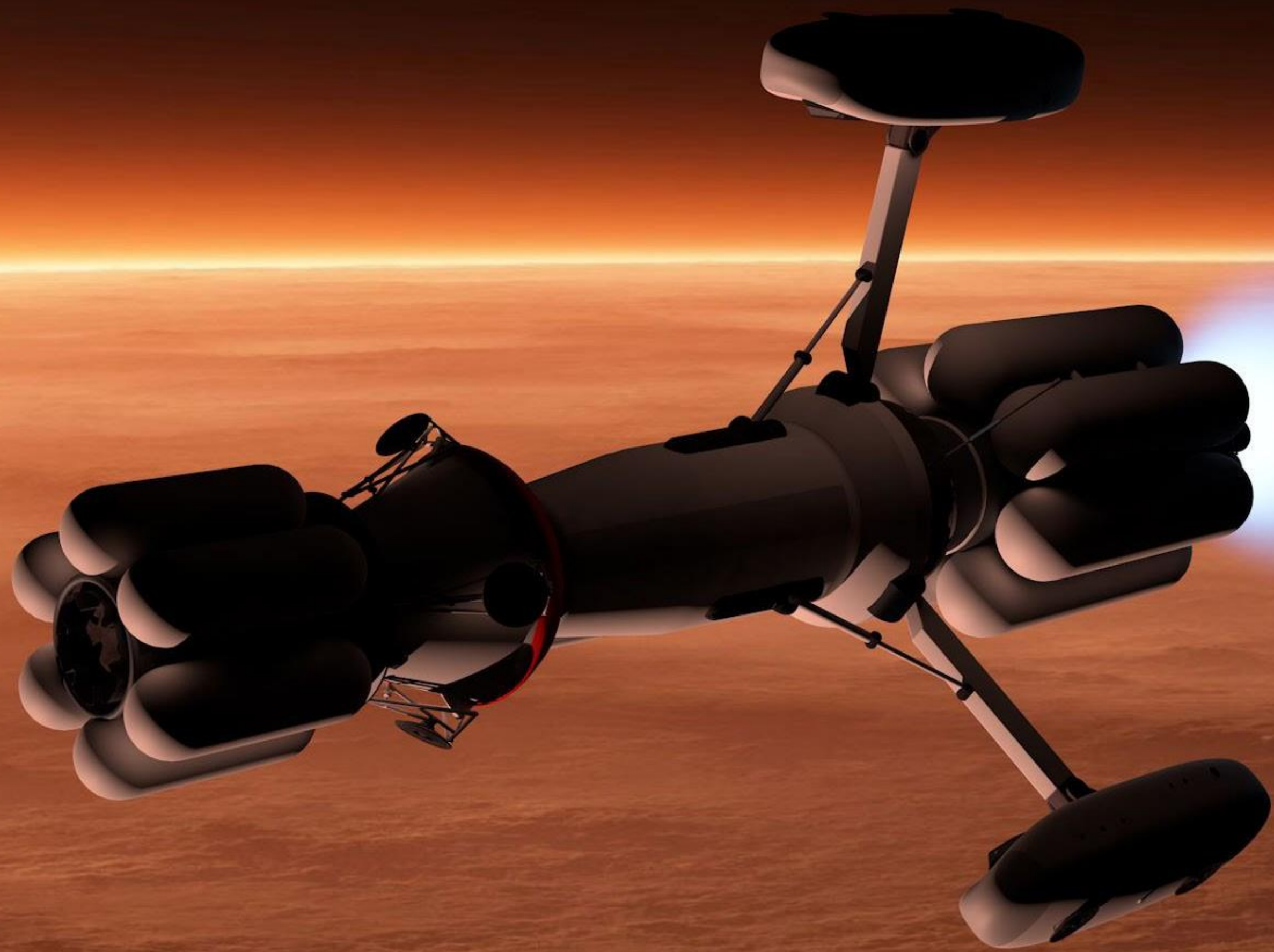








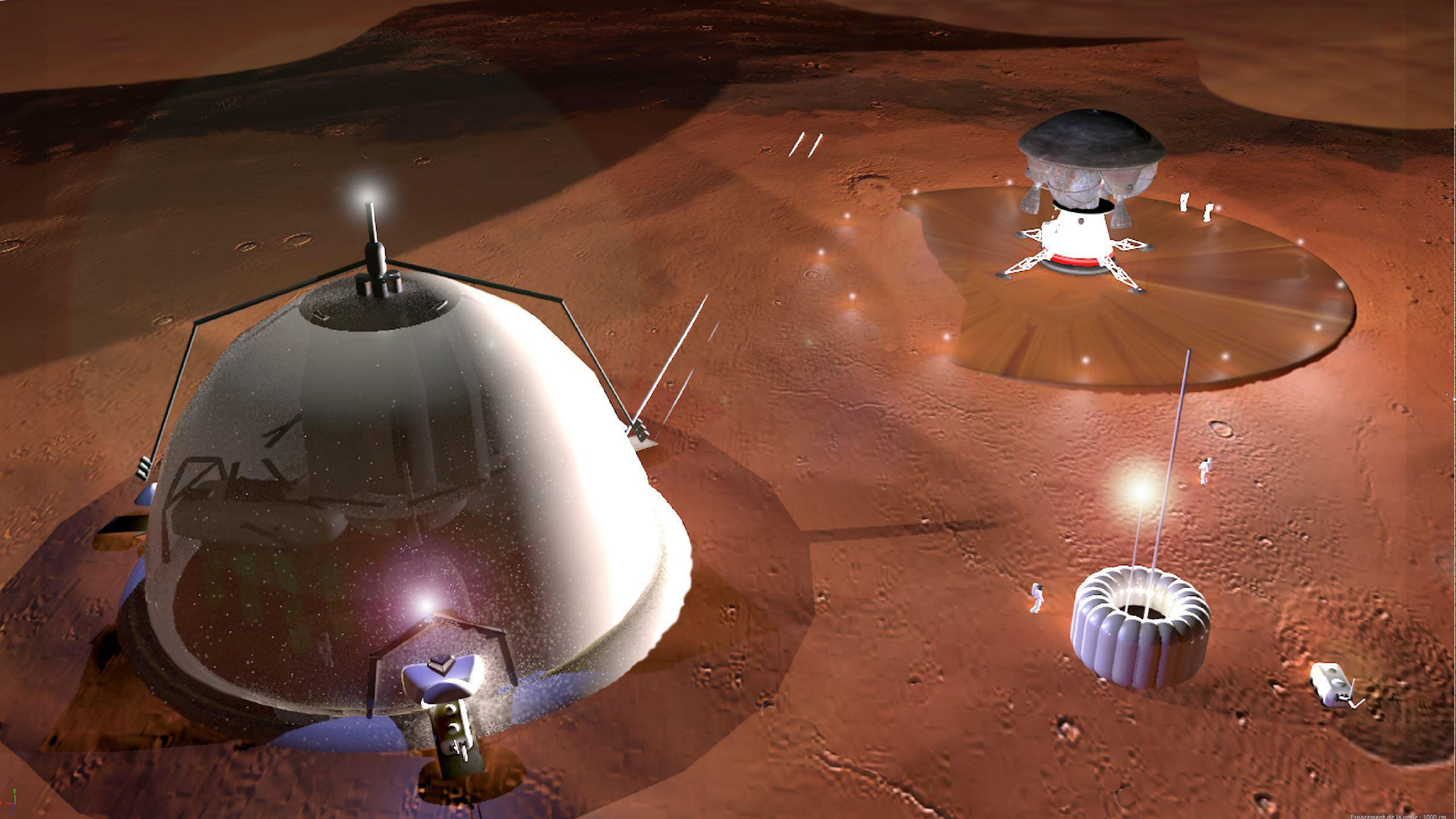






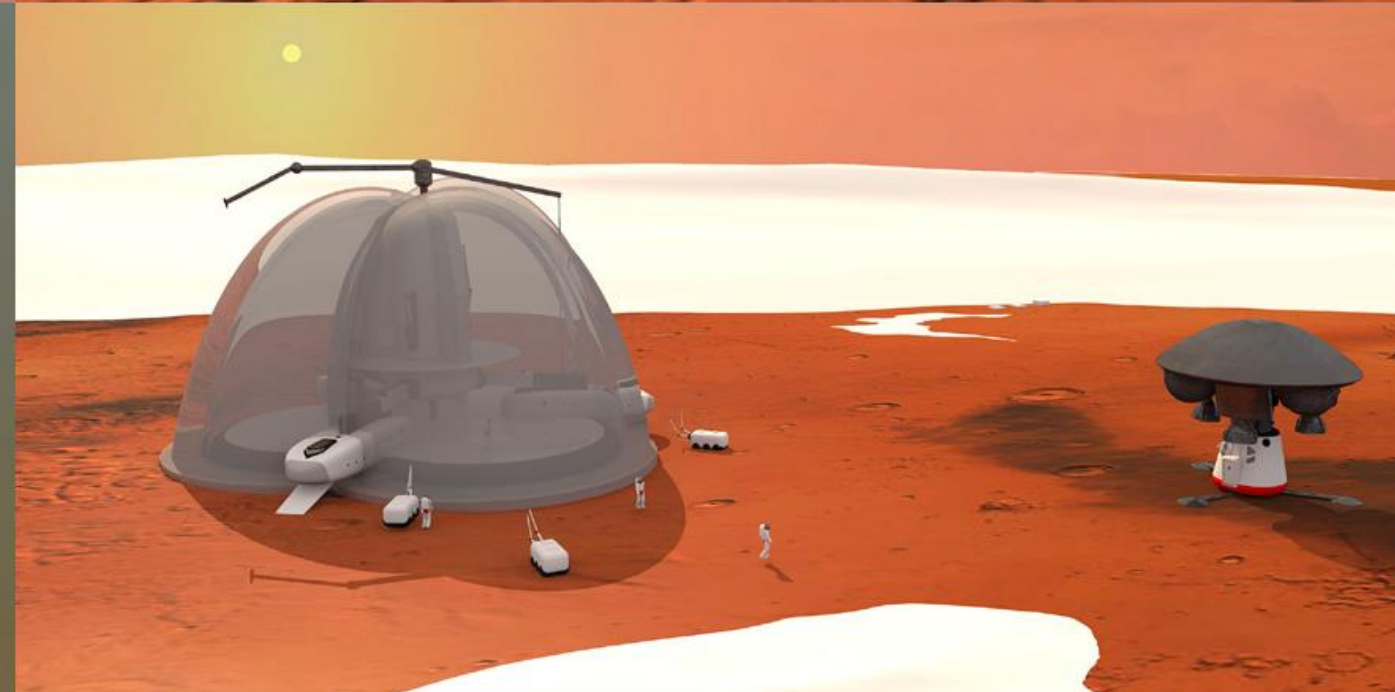
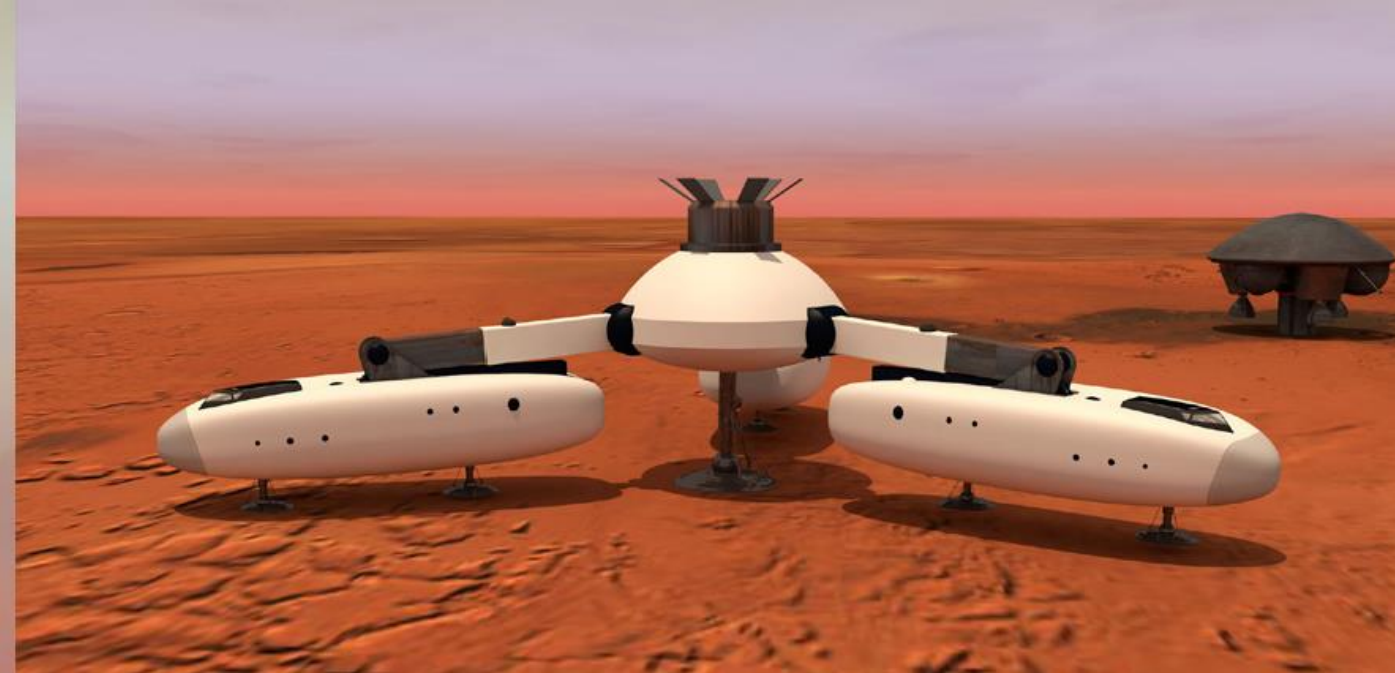
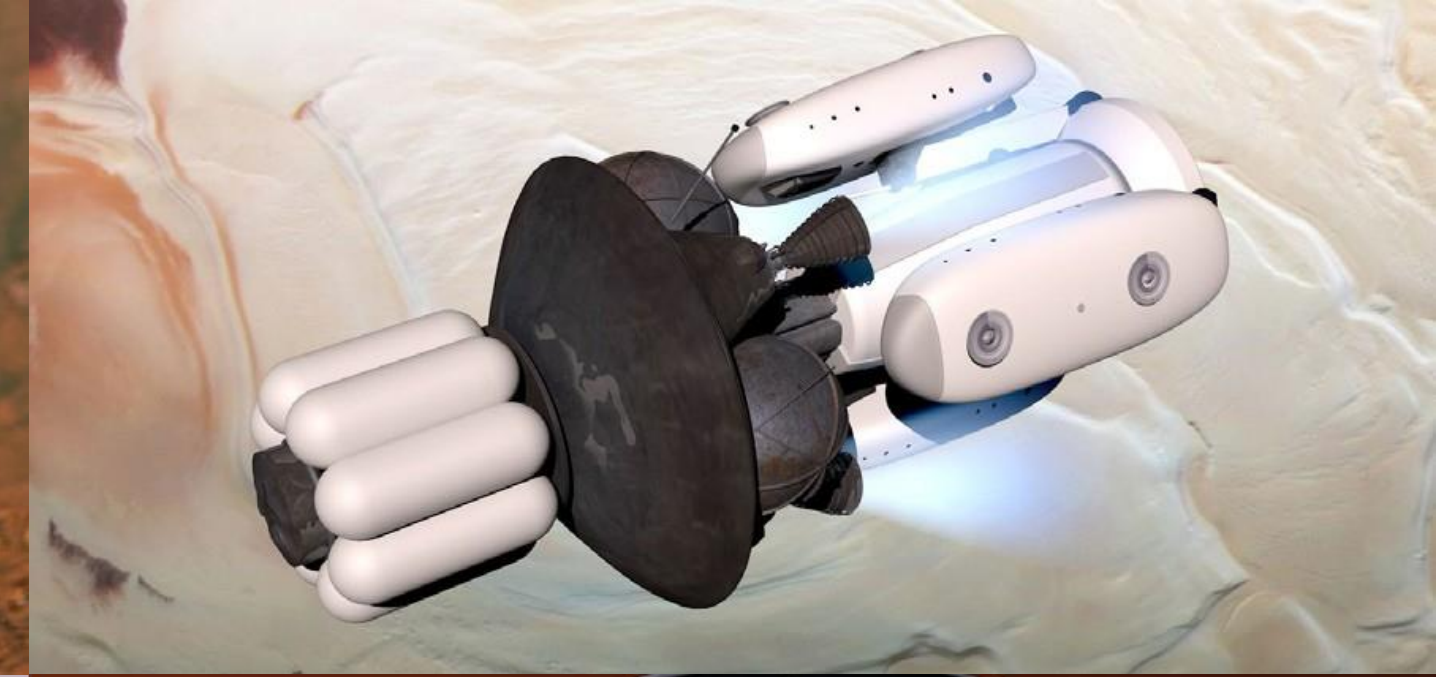
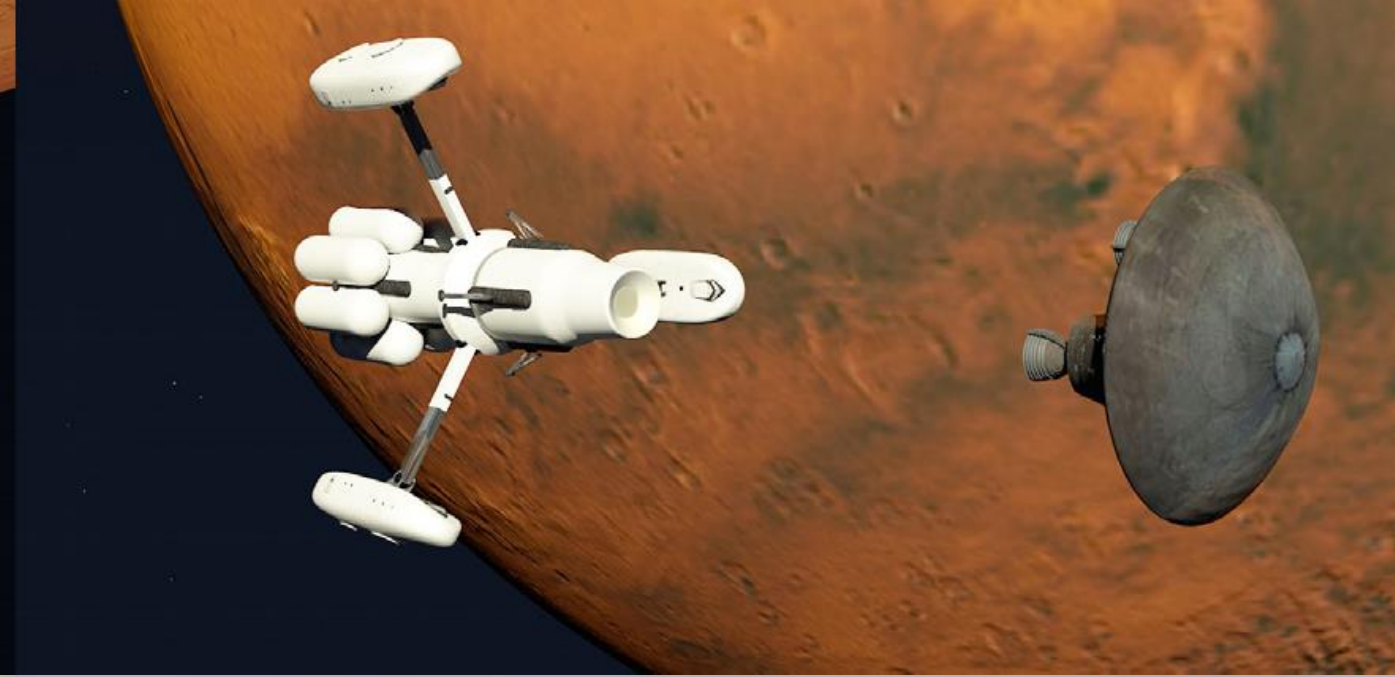
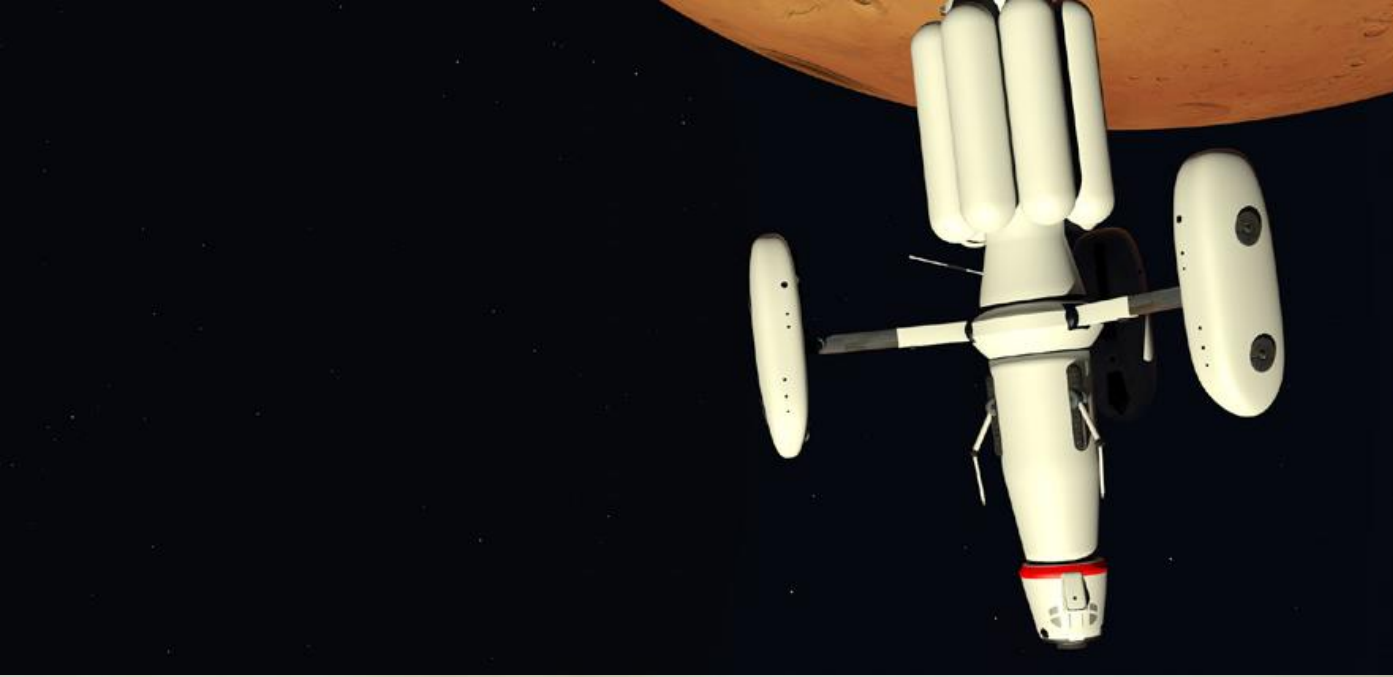






Conclusions

- Identification of potential in-situ resource utilization strategy based on ethylene and ice
- Recommendation to develop deployable structures
- Identification of limits to autonomy from Earth and follow-up recommendation to include greenhouse and energy production in-situ testing
- Recommendation to develop Mars crane system



Anne-Marlene Rüede
annemarlene.ruede@yahoo.com